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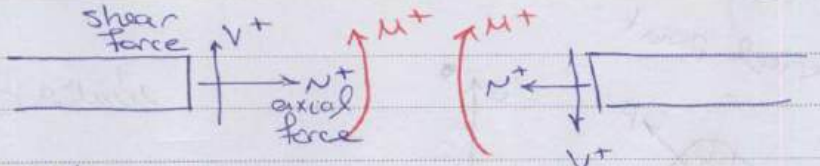
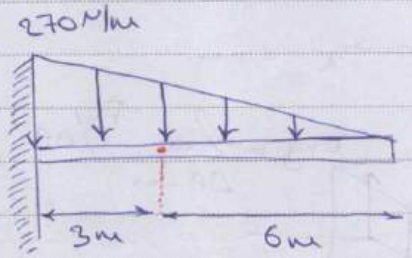
* مقاومت مصالح *

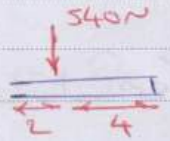




مثال:

$\frac{w}{6} = \frac{270}{9} \Rightarrow w = 180N$

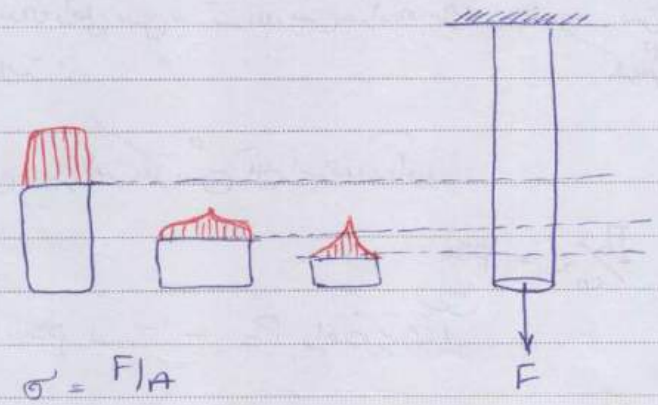
راحت تر ②



$N = 0$

$V = 540N$

$M = -1080 \, Nm$



$\sigma = F/A$

normal stress

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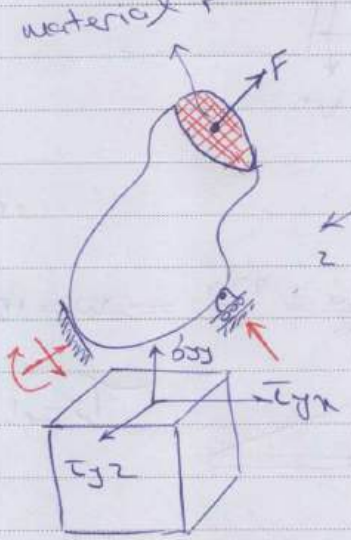
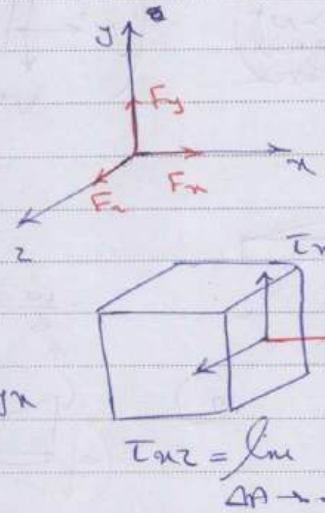
تنش عمودی σ
تنش برشی τ

material point

تنش در یک نقطه





$$\tau_{xy} = \lim_{\Delta A \to 0} F_y/\Delta A$$

$$\sigma_{xx} = \lim_{\Delta A \to 0} \frac{F_x}{\Delta A}$$

$$\tau_{xz} = \lim_{\Delta A \to 0} \frac{F_z}{\Delta A}$$

نقطه‌ای است بزرگ‌تر از اتم ولی خیلی کوچک‌تر از اجسام.

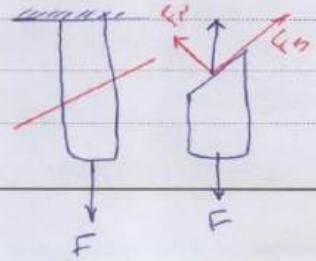
کوچکترین قطعه جسم است که خواص جسم را حفظ کند. material point

stress	تنش	خودنمودهای داخلی تنش نیستند بلکه تنش مؤلفه‌های تنش داخلی است.
strain	کرنش	

کرنش چیزی است که تنش آن را به وجود آورد.

$N/m^2 = Pa$ $\quad lb/in^2 = psi$

psi نسبت به Pa بزرگتر است



$\sigma = \frac{F_n}{A_0}$ تنش محوری

$\tau = \frac{F_s}{A_0}$ تنش برشی

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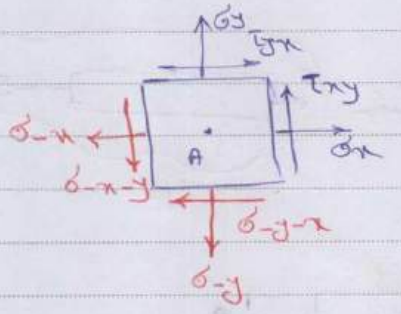
$$\text{تنش} = \begin{pmatrix} \sigma_x & \tau_{xy} & \tau_{xz} \\ \tau_{yx} & \sigma_y & \tau_{yz} \\ \tau_{zx} & \tau_{zy} & \sigma_z \end{pmatrix}$$

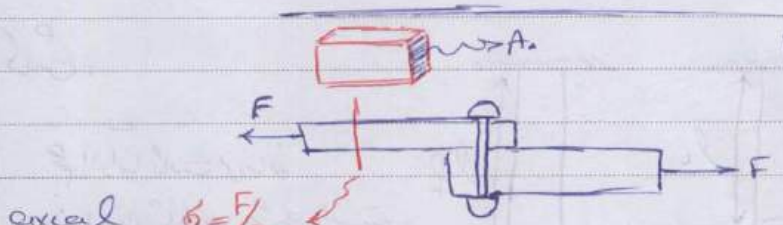
مؤلفه‌های تنش در حالت کلی (3 بعدی)

در حالت دو بعدی تا حدی نشون



$$\begin{cases} \sum F_x = 0 \\ \sum F_y = 0 \\ \sum M_{z/A} = 0 \end{cases} \Rightarrow \tau_{xy} = \tau_{yx}$$

$$\text{تنش} = \begin{pmatrix} \sigma_x & \tau \\ \tau & \sigma_y \end{pmatrix}$$



bearing stress
F/td

$\tau = F/A.$ single shear shearing stress

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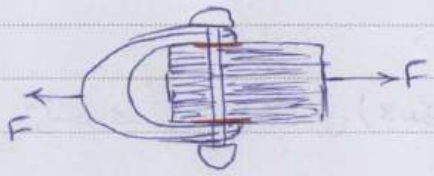
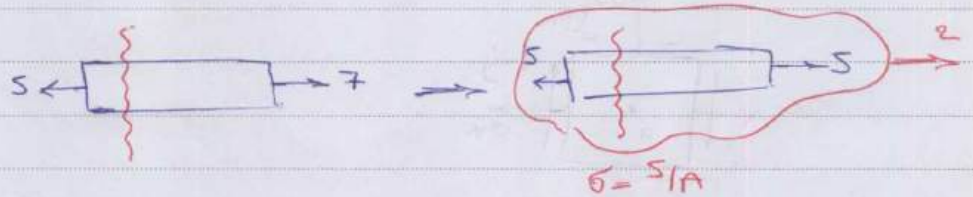
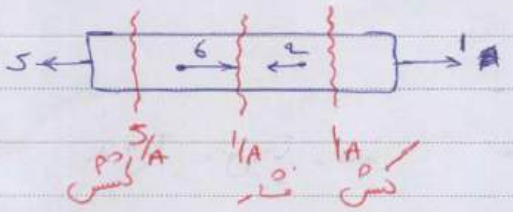
double shear

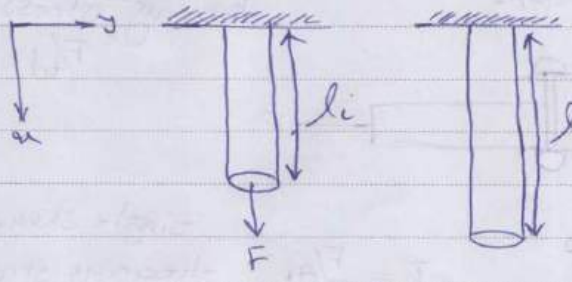


$$\tau = \frac{F/2}{A} = \frac{F}{2A}$$





$$\text{تنش مجاز (allowable stress)} = \frac{\text{تنش تسلیم (کاری)}}{\text{safty factor}}$$



فشار:

حجم اولیه ثابت نمی‌ماند

جهت باقی ایجاد شده دانسیته

تغییر می‌کند

$$\text{strain} = \frac{l_f - l_i}{l_i} = \varepsilon_x > 0$$

طول نمونه زیاد می‌شود

$$\frac{d_f - d_i}{d_i} = \varepsilon_y < 0$$

قطر نمونه کم می‌شود

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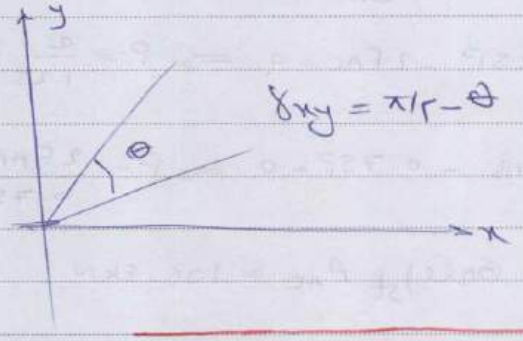
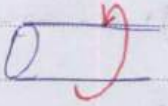
کلاس حل تمرین 12:30-13:30 سه‌شنبه

تنش = $\begin{pmatrix} \sigma_x & \tau_{xy} & \tau_{xz} \\ \tau_{yx} & \sigma_y & \tau_{yz} \\ \tau_{zx} & \tau_{zy} & \sigma_z \end{pmatrix}$ 3 بعدی

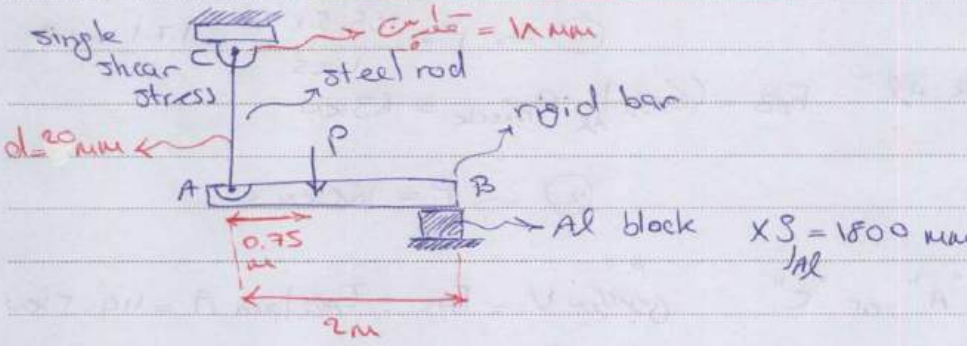
تنش = $\begin{pmatrix} \sigma_x & \tau_{xy} \\ \tau_{yx} & \varepsilon_y \end{pmatrix}$ 2 بعدی



$\gamma_{xy} = \pi/2 - \theta$



مثال:



$A_{S_{Al}} = 1800\ mm^2$

$\sigma_F)_{st} = 680\ MPa$ (Failure)

$P_{MAX} = ?$ مطلوب

$\sigma_F)_{Al} = 70\ MPa$

$S.F = 2$

$\tau_F)_{pin} = 900\ MPa$

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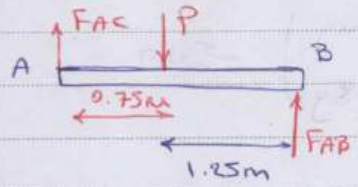
$\sigma_{All})_{st} = \frac{680}{2} = 340\ MPa$ $\quad \sigma_{All})_{Al} = 35\ MPa$

$\tau_{all})_{pin} = 450\ MPa$



$\sum M_B \circlearrowleft + = 0 \Rightarrow 1.25P - 2F_{AC} = 0 \Rightarrow P = \frac{2}{1.25} F_{AC}$ (1)

$\sum M_A \circlearrowleft + = 0 \Rightarrow 2F_{AB} - 0.75P = 0 \Rightarrow P = \frac{2F_{AB}}{0.75}$ (2)

"Rod AC" $\quad F_{AC} = \sigma_{All})_{st} A_{AC} = 106.8\ kN$

(1) $\Rightarrow P = \frac{106.8 \times 2}{1.25} = 171\ kN$ *

"Block Al" $\quad F_{AB} = (\sigma_{All})_{Al} A_{Block} = 63\ kN$

(2) $\Rightarrow P = 168\ kN$ *

Pin "A" or "C" $\quad$ برشی $V = F_{AC} = \tau_{All})_{pin} A = 114.5\ kN$

(1) $\Rightarrow P = \frac{11.4(2)}{1.25} = 183\ kN$ *

* $\Rightarrow P_{max} = 168\ kN$

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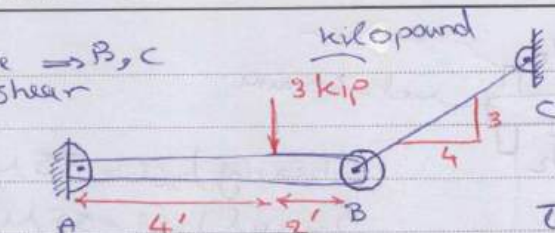
single shear $\Rightarrow$ B, C

A $\Rightarrow$ double shear



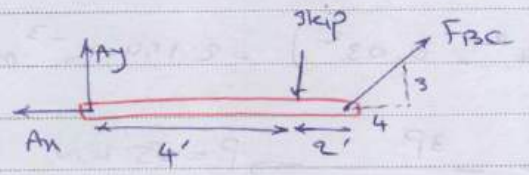
kilopound : ثقل

ft ، inch

$\tau_{All/pin} = 12.5$ ksi $\left(\frac{\text{kpound}}{\text{sq. inch}}\right)$

$\sigma_{All/CB} = 16.2$ ksi

کوچکترین قطر سن و قطر میله BC را تعیین کنید تا $\left(\frac{1}{16}''\right)$



* دقت: نیروی وارد به سن A برابر با برآیند نیروی A است نه نیروی A در راستای عمود

$\sum M_B \circlearrowleft + = 0 \Rightarrow 6A_y = 3\times 2 \Rightarrow A_y = 1$ kip

$\sum M_A \circlearrowleft + = 0 \Rightarrow 3\times 4 = 6B_y \Rightarrow B_y = 2$ kip

$F_{BC} = 5/3\, B_y = 3.33$ kip

$A_x = 4/5\, F_{BC} = 2.67$ kip

$V_A = \sqrt{1^2 + (2.67)^2} = 2.85$ kip

$A_A = \dfrac{2.85/2}{\tau_{All}} \Rightarrow d_A = 0.381''$

$A_C = \dfrac{3.33}{12.5} \Rightarrow d_B = 0.625''$

$A_{BC} = \dfrac{F_{BC}}{\sigma_{All/BC}} = \dfrac{3.33}{16.2} \Rightarrow d_{BC} = 0.512''$

choose $0.5625''$

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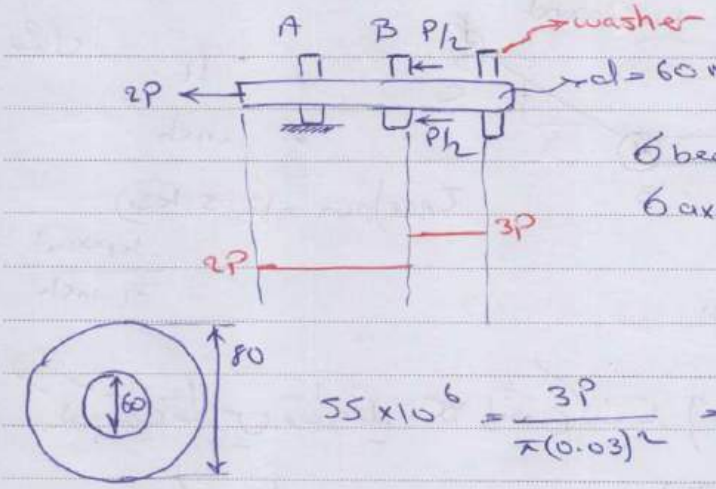


: du

$\sigma_{bearing\,all} = 75\ MPa$

$\sigma_{axial\,all} = 55\ MPa$

$$55\times10^{6} = \frac{3P}{\pi(0.03)^2} \rightarrow P = 51.8\ kN$$

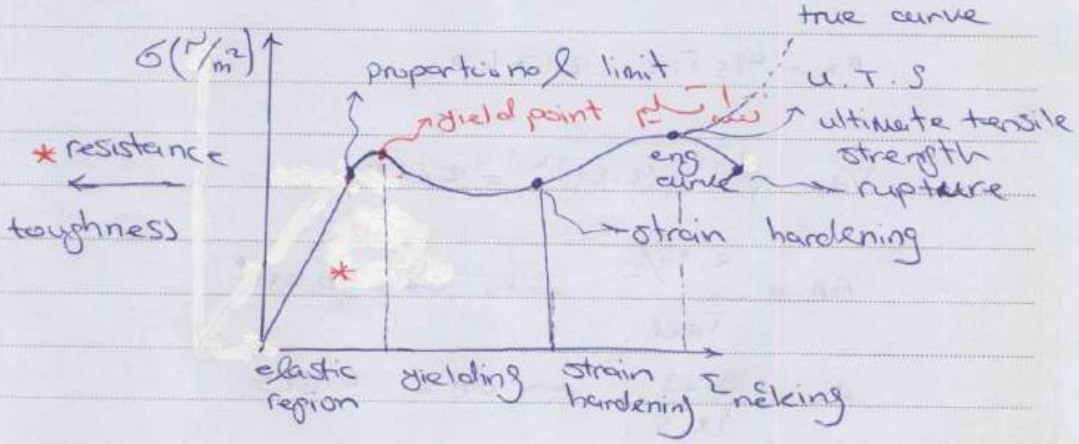
$$A_b = \pi(0.04^2 - 0.03^2) = 2.199\times10^{-3}\ m^2$$

$$2.199\times10^{-3} = \frac{3P}{75\times10^{6}} \rightarrow P = 55\ kN$$

$$P_{max} = 51.8\ kN$$

---

ductile materials



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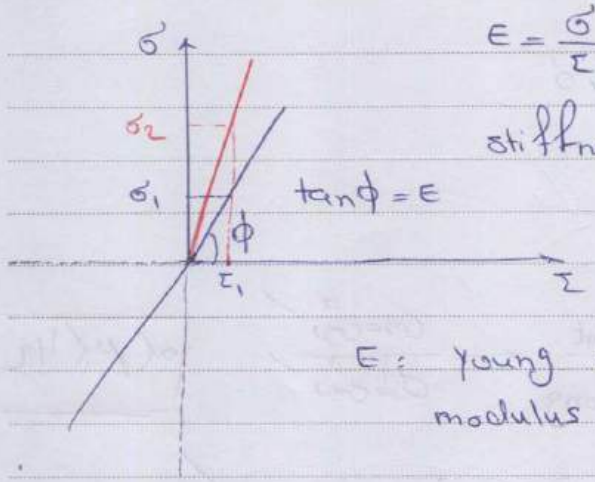


$E = \frac{\sigma}{\varepsilon} \rightarrow \boxed{\sigma = E\,\varepsilon}$ Hook's Law

stiffness

E: young modulus
modulus of elasticity

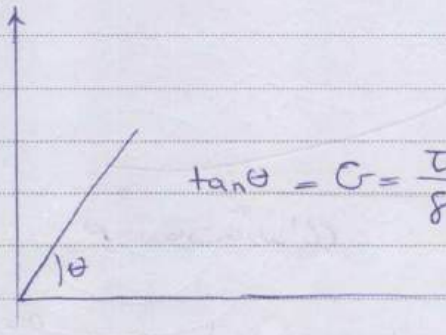
$E \equiv \frac{\sigma}{\varepsilon}$ N/m^2

$E_c = E_t$ ← برای اکثر مصالح

strain energy ⟵ انرژی مصرف شده برای تغییر حجم ΔV

$\Delta U = \frac{1}{2}\sigma\varepsilon\,\Delta V = \frac{1}{2}\frac{F}{L^2}\frac{L}{L}L^3 = F.L$

U_0 = انرژی کرنشی واحد حجم $= \frac{\Delta U}{\Delta V} = \frac{1}{2}\sigma\varepsilon = \frac{1}{2}\sigma\frac{\sigma}{E} = \frac{1}{2}\frac{\sigma^2}{E}$ (سطح زیر نمودار $\sigma-\varepsilon$)



$\tan\theta = G = \frac{\tau}{\gamma} \rightarrow \boxed{\tau = G\gamma}$

shear elastic modulus

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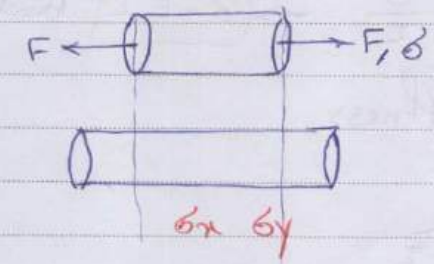


ضریب پواسون $\mu = -\frac{\varepsilon_{lat}}{\varepsilon_{long}} = -\frac{\text{کرنش عرضی}}{\text{کرنش طولی}}$

$0 < \mu < 1/2$

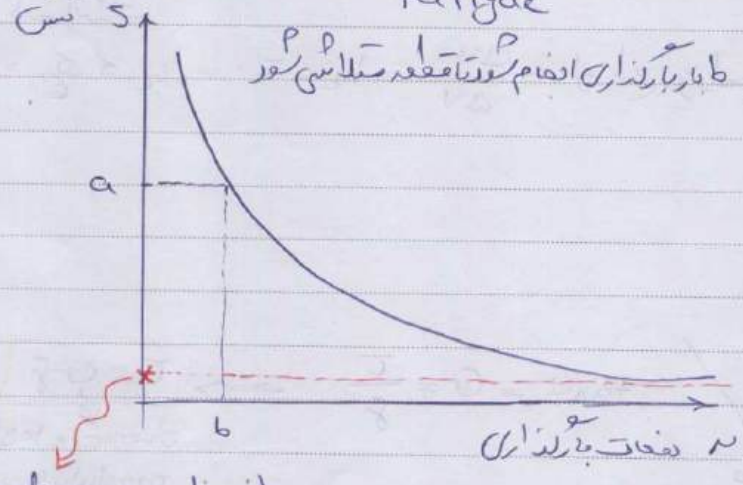
poisson ratio

برای مواد غیرقابل تراکم (حجم ثابت) $\Rightarrow \mu = 1/2$

$$G = \frac{E}{2(1+\mu)}$$

Fatigue

تنش S



بارگذاری تنش a را باید b بار بارگذاری انجام شود تا قطعه متلاشی شود

* S-N curve *

Endurance limit

برای کلیه قطعه تنش‌های زیر endurance limit دیگر خرابی نمی‌شود

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(Creep خزش) کرنش

تنش ثابت

فوراً بعد از کشش رسم شده

کرنش بلافاصله بعد از اعمال تنش

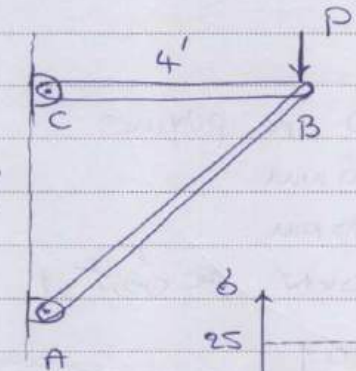
$\varepsilon = \frac{1}{E}\sigma$

زمان

کرنش

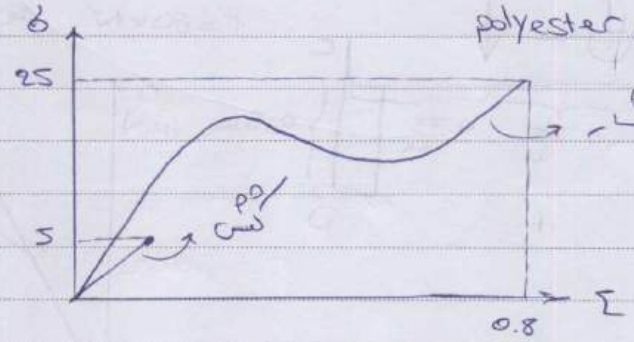
زمان

مثال :



سطح AB = $1.5 inch^2$

سطح BC = $4 inch^2$



از جنس polyester

تنش

failure از قبل P_{MAX} = ?

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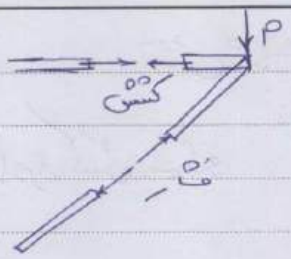


$\uparrow \sum F_y = 0 \Rightarrow \frac{3}{5} F_{AB} - P = 0$

$\Rightarrow F_{AB} = 1.667P$

$\sum F_x = 0 \Rightarrow \frac{4}{5} F_{AB} = F_{BC}$

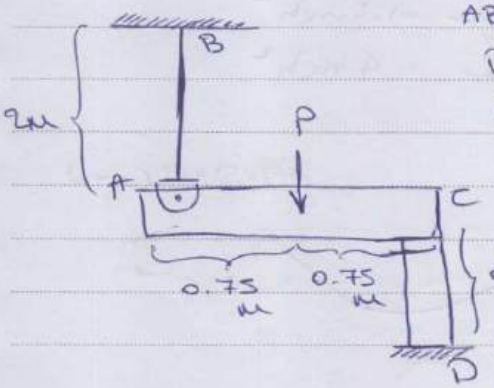
$\Rightarrow F_{BC} = \frac{4}{5}(1.667P) = 1.333P$

$\sigma_{BC} = \frac{F_{BC}}{A_{BC}} \Rightarrow 5 = \frac{F_{BC}}{4} \Rightarrow F_{BC} = 20\,kip$

$\Rightarrow P = \frac{20}{1.333} = 15\,kip$

$25 = \frac{F_{AB}}{\frac{A_{AB}}{1.5}} \Rightarrow F_{AB} = 37.5\,kip \Rightarrow P = \frac{37.5}{1.667} = 22.5\,kip$

$\Rightarrow \boxed{P_{max} = 15\,kip}$

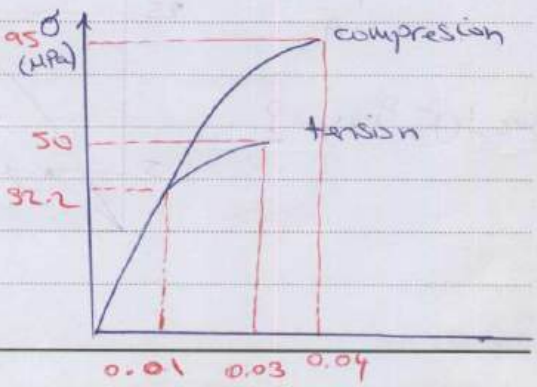


AB & CD are polymers

$D_{AB} = 60\,mm$

$D_{CD} = 80\,mm$

$P = 80\,kN$ AC تغییر طول =؟



PAPCO

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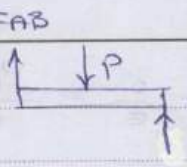


$F_{AB} = F_{CD} = P/2 = 40kN$

$\sigma_{AB} = \frac{F_{AB}}{A_{AB}} = \frac{40\times10^3}{\frac{\pi}{4}(0.04)^2} = 31.83$ MPa $<$ 32.2 MPa $= \sigma_{yield}$

$\sigma = E\varepsilon$ استفاده کرد proportional limit در

$E = \frac{32.2(10^6)}{0.01} = 3.22$ GPa

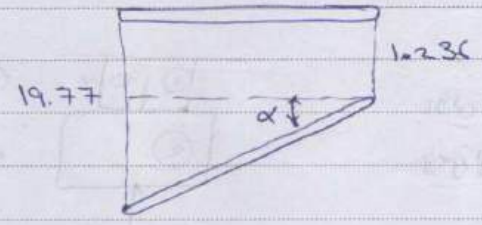
$\varepsilon_{AB} = \frac{\sigma_{AB}}{E} = \frac{31.83\times10^6}{3.22\times10^9} = 0.0098$

$\delta_{AB} = \varepsilon_{AB} L_{AB} = 0.0098 \times 2000 = 19.77$ mm

$\sigma_{CD} = \frac{F_{CD}}{A_{CD}} = \frac{40\times10^3}{\frac{\pi}{4}(0.08)^2} = 7.958$ MPa in the proportional limit.

$\varepsilon_{CD} = \frac{\sigma_{CD}}{E} = \frac{7.958\times10^6}{3.22\times10^9} = 0.00247$

$\delta_{CD} = \varepsilon_{CD} L_{CD} = 0.00247 \times (500) = 1.236$ mm



$\tan\alpha = \frac{19.77 - 1.236}{1500}$

$\rightarrow \alpha = 0.708°$

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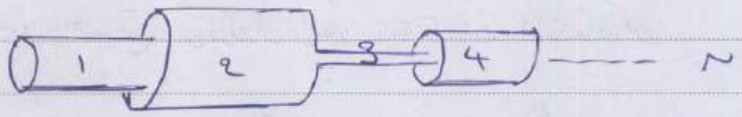
$$\sigma = E\varepsilon \Rightarrow \varepsilon = \frac{\sigma}{E} = \frac{P/A}{E} \Rightarrow \frac{\Delta l}{l} = \frac{P/A}{E}$$

$$\Rightarrow \Delta l = \frac{Pl}{AE}$$

$$\Delta l = \sum_{i=1}^{n} \frac{P_i l_i}{A_i E_i}$$

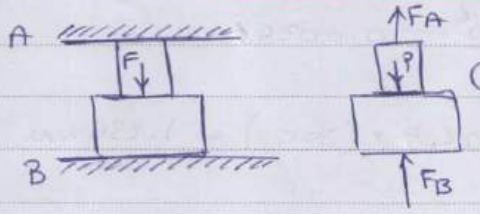


$$\Delta l = \int_0^l \frac{F(x)\,dx}{A(x)\,E(x)}$$

E معمولاً ثابت است و صرفاً تابعی از x نیست مگر اینکه مثلاً قطعه از 3 جنس مختلف باشد:

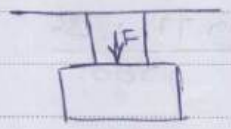
$$\sum_{i=1}^{3} \frac{1}{E_i} \int_0^{l_i} \frac{F(x)\,dx}{A(x)}$$

مثال:

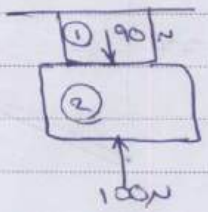


از تعادل نیروها در راستای قائم $F_A + F_B = P$

تغییر طول متناسب $\Delta l_1 + \Delta l_2 = 0$

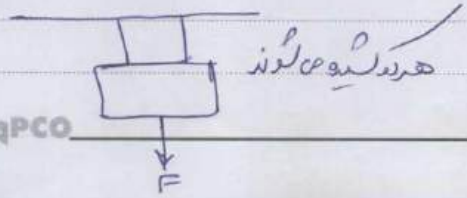


بالایی کشیده می‌شود
پایینی تغییری نمی‌کند



تنش $\sigma_1 = \frac{10}{A_1}$

تنش $\sigma_2 = \frac{100}{A_2}$



هردو کشیده می‌شوند

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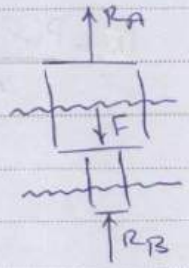
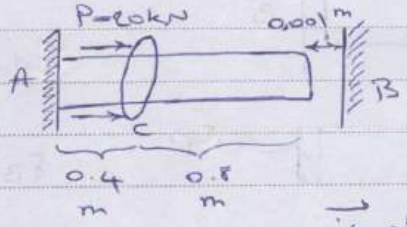


$$\sum F_y = 0 \Rightarrow R_A - F + R_B = 0 \quad (1)$$

$$\delta = \frac{R_A l_1}{A_1 E_1} - \frac{R_B l_2}{A_2 E_2} = 0 \quad (2)$$

(1), (2) $\Rightarrow R_A$, R_B

مثال:

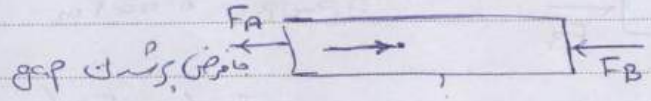


$E = 200\,GPa$

$d = 0.005\,m$

راه حل: تغییر طول AB را برابر با 0.001 در نظر گرفته و سپس بزرگتر از 0.001 است یا کوچکتر

BC , AC بررسی



فرض پرشک جمع

$$F_A + F_B = 20 \times 10^3\,N \quad (1)$$

$$\Delta l = \frac{F_A L_{AC}}{AE} - \frac{F_B L_{BC}}{AE} = 0.001$$

$$\Rightarrow 0.4 F_A - 0.8 F_B = 3927 \quad (2)$$

(1), (2) $\Rightarrow F_A = 16.6\,kN$

$F_B = 3.39\,kN$

P4PCO

فرض پرشک جمع زده شده چون F_B مثبت شد

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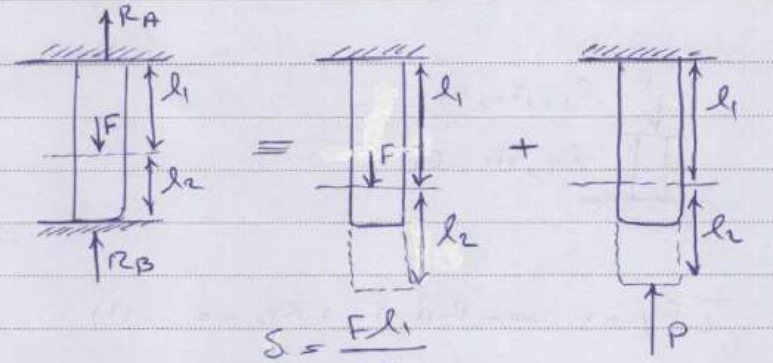
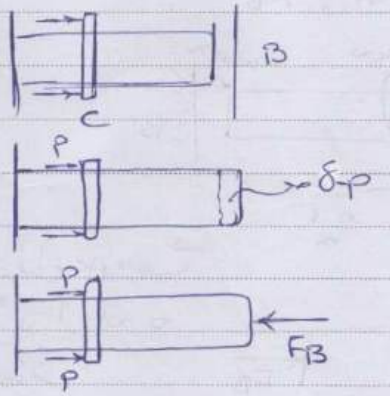


$$\delta = \frac{P(l_1+l_2)}{AE}$$

$$\delta = \frac{Fl_1}{AE}$$

$$\frac{Fl_1}{AE} = \frac{P(l_1+l_2)}{AE}$$

مثال دوم:



$$\delta_P = \frac{PL_{AC}}{AE} = 0.002\,m$$

$$\delta_B = \frac{F_B L_{AB}}{AE}$$

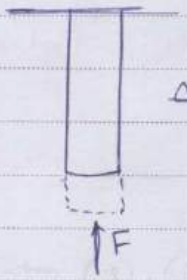
$$\delta_P - \delta_B = 0.001\,m$$

$\Rightarrow F_A$ ✓ , F_B ✓

$$\delta_T = \alpha(\Delta T)l$$

$$\delta_F = \frac{Fl}{AE}$$

$$\alpha(\Delta T)l = \frac{Fl}{AE}$$



$$\rightarrow F = \alpha \Delta T A E$$

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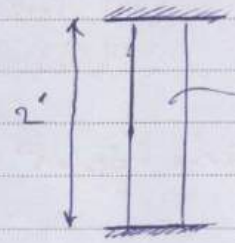
: due



$T_1 = 60\ ^\circ F \quad T_2 = 120\ ^\circ F$

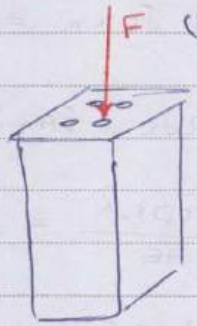
$$F = 6.6\times10^{-6}\ \frac{1}{^\circ F}\,(120-60)\,^\circ F\,(0.5)^2\times29\times10^3$$

$$= 2.87\ \text{kip}$$

$$\sigma = \frac{2.87}{(0.5)^2} = 11.5\ \text{ksi}$$

---

composites



(1) $F = f_B + f_S \qquad \delta_B = \delta_S$

(2) $\dfrac{f_B \ell}{A_B E_B} = \dfrac{f_S \ell}{A_S E_S}$

(1), (2) $\Rightarrow f_B$ ✓, f_S ✓

: due

bars made of same material with same XS

find the forces in each bar

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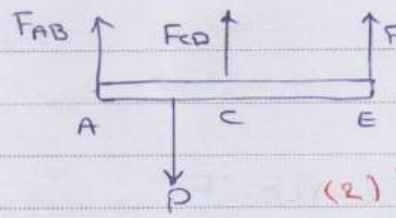
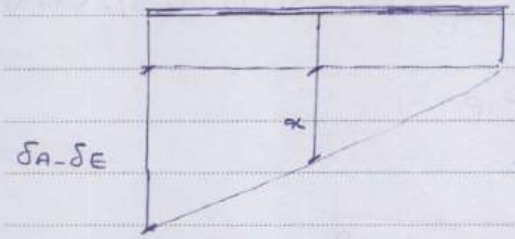


(1) $\sum M_A = 0 \Rightarrow F_{EF}(2d) + F_{CD}(d) - P(d/2) = 0$

(2) $\sum F_y = 0 \Rightarrow F_{AB} + F_{CD} + F_{EF} = P$



$$\frac{x}{d} = \frac{\delta_A - \delta_E}{2d} \Rightarrow x = \tfrac{1}{2}(\delta_A - \delta_E)$$

$$\delta_C = \delta_E + x = \frac{\delta_A + \delta_E}{2}$$

$$\Rightarrow 2\delta_C = \delta_A + \delta_E$$

$$\Rightarrow \frac{2F_{CD}\,l}{AE} = \frac{F_{AB}\,l}{AE} + \frac{F_{EF}\,l}{AE}$$

$$\Rightarrow (3)\ 2F_{CD} = F_{AB} + F_{EF}$$

$$F_{CD} = P/3 \qquad F_{EF} = P/12 \qquad F_{AB} = 7/12\,P$$

$\delta = \frac{\delta l^2}{2E}$

$\delta = \frac{\delta_A l (l/2)}{AE}$ $\delta_A l$ l A, δ $l/2$ P

$$\delta = \int_0^l \frac{f(x)\,dx}{AE}$$

$\delta_A x + P = f(x)$ x $\delta_A x$ P

$$= \frac{1}{AE}\int_0^l (\delta_A x + P)\,dx$$

$$= \frac{1}{AE}\left(\frac{\delta_A x^2}{2} + Px\Big|_0^l\right)$$

$$= \frac{\delta l^2}{2E} + \frac{Pl}{AE}$$

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حل تمرین مقاومت ۱

فصل ۱ : 37 _ 33 _ 65 _ 43 _ 18

Sec 1 ⇒ (2-15), (2-19), (2-21), (2-23), (2-30)

Sec 2 ⇒ (2-37), 2-41, 2-43, (2-45), (2-46)

Sec 3 ⇒ 2-49, 2-55, (2-60)

Sec 4 ⇒ 2-73, 2-74, 2-83, 2-84

Sec 5 ⇒ (1-9), (1-15), (1-20), (1-28)

Sec 6 ⇒ 2-66, 2-75, 2-76

Sec 7 ⇒ (3-11), (3-21), 3-48, 3-50, 3-61, 3-63

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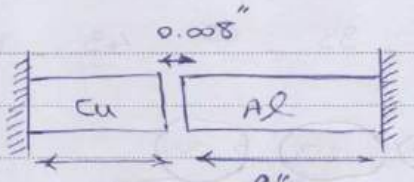
پیش‌نیاز ۱۷ انبساط حرارتی - سرکلاس



$T_1 = 60\ ^\circ F$

$d = 1.25"$

$\alpha_{Al} = 13 \times 10^{-6}\ 1/F$

$E_{Al} = 10 \times 10^3\ ksi$

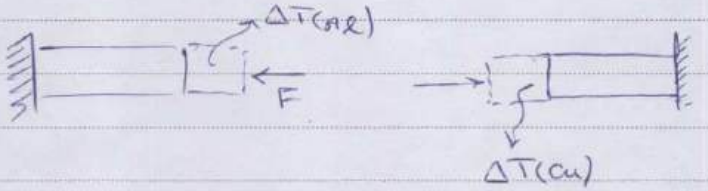
$\alpha_{Cu} = 9.4 \times 10^{-6}\ 1/F$

$E_{Cu} = 18 \times 10^3\ ksi$

① $T_2 = ?$ برای P_{max} شود

② $T_2 = 200^\circ F \Rightarrow \sigma = ?$

③ $T_2 = 300^\circ F \Rightarrow L_{(Al)} = ?$



① $\delta_T = 0.008 = \alpha_{Al} \underbrace{\Delta T}_{(T_2-60)} L_{Al} + \alpha_{Cu} \Delta T L_{Cu} \Rightarrow T_2 = 116\ ^\circ F$

② $0.008 = (\delta T)_{Cu} - (\delta F)_{Cu} + (\delta T)_{Al} - \underbrace{(\delta F)_{Al}}_{(\delta F)_{Cu}}$

$= 9.4 \times 10^{-6}(200-60)4 - \dfrac{F(4)}{\frac{\pi}{4}(1.25)^2 18 \times 10^3}$

$+ 13 \times 10^{-6}(200-60)8 - \dfrac{F(8)}{\frac{\pi}{4}(1.25)^2 (10 \times 10^3)}$

$\Rightarrow F = 14.195\ kip$

$\sigma_{Al} = \sigma_{Cu} = \dfrac{14.195}{\frac{\pi}{4}(1.25)^2} = 11.6\ ksi$

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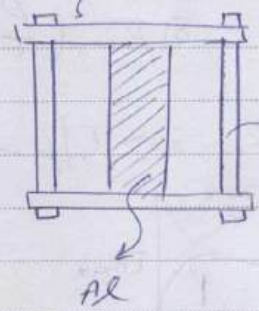
③ $T_c = 300$ ... $\Rightarrow F = 31.194$ kip

$$\delta_{Al} = (\delta_T)_{Al} - (\delta_F)_{Al} = 0.00426''$$

$$L_{Al} = 8 + 0.00426 = 8.00426''$$

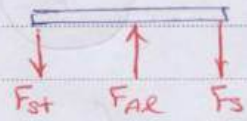
this figure makes no sense.

$$\begin{cases} d_{Al} = 50\ mm \\ d_{st} = 10\ mm \end{cases} \qquad \begin{cases} l_{Al} = 100\ mm \\ l_{st} = 150\ mm \end{cases}$$

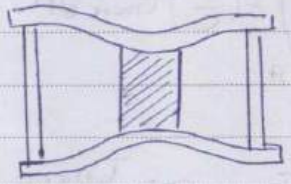


$$\begin{cases} \alpha_{st} = 17\times10^{-6}\ 1/°C \\ \alpha_{Al} = 26\times10^{-6}\ 1/°C \end{cases} \qquad \begin{cases} E_{st} = 193\times10^{9}\ Pa \\ E_{Al} = 44.7\times10^{9}\ Pa \end{cases}$$

$\Delta T = 110$



$$\Rightarrow \begin{cases} 2F_{st} = F_{Al} \\ \delta_{Al} = \delta_{st} \end{cases}$$



$$(\delta_T + \delta_F)_{st} = (\delta_T - \delta_F)_{Al}$$

$$17\times10^{-6}\times0.15\times110 + \frac{F(0.15)}{193(10^9)(2)\pi/4\,(0.01)^2} = 26\times10^{-6}\times0.1\times110$$

$$- \frac{F(0.1)}{44.7(10^9)\frac{\pi}{4}(0.05)^2}$$

$$\Rightarrow F = 904\ N$$

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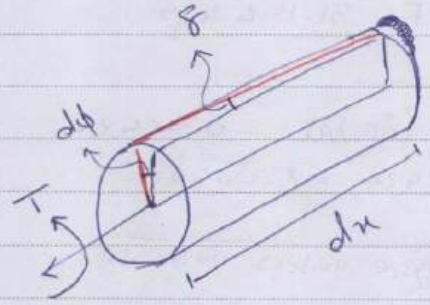
77-60-21-17 (2)

فصل 3 :



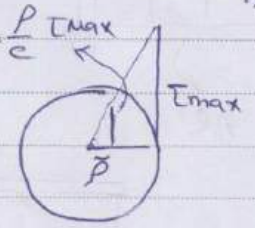
$$c\,d\phi = \delta\,dx$$

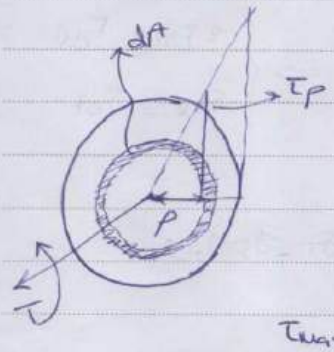
$$\Rightarrow \delta = \frac{c\,d\phi}{dx}$$

$$\delta|_\rho = \left(\frac{\rho}{c}\right)\delta_{max}$$

$$\tau = \delta G \Rightarrow \tau|_\rho = \frac{\rho}{c}\,\tau_{max}$$

$$\tau|_\rho = \frac{\rho}{c}\,\tau_{max}$$





$$T = \int_A \rho(\tau_\rho\,dA) = \int_A \rho\left(\frac{\rho}{c}\right)\tau_{max}\,dA$$

$$= \frac{\tau_{max}}{c}\int_A \rho^2\,dA = \frac{\tau_{max}}{c}J \longrightarrow \tau_{max} = \frac{Tc}{J}$$

$$\tau_\rho = \frac{T\rho}{J}$$

$$J = I_x + I_y = \int \underbrace{(x^2+y^2)}_{\rho^2}\,dA$$

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$\bar{J}$ برای سطح دایره‌ای $\bar{J} = \int_A \rho^2 dA = \int_0^c \rho^2 (2\pi\rho d\rho) = 2\pi \int_0^c \rho^3 d\rho$

$$\Rightarrow \boxed{\bar{J} = \frac{\pi}{2} c^4}$$

برای دایره توخالی $\bar{J} = 2\pi \int_{R_i}^{R_o} \rho^3 d\rho = \boxed{\frac{\pi}{2}(R_o^4 - R_i^4)}$

$$c\,d\phi = \delta\,dx \Rightarrow \rho\,d\phi = \delta_\rho\,dx$$

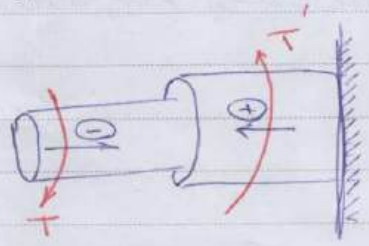
$$\Rightarrow d\phi = \frac{\delta_\rho}{\rho}\,dx$$

$$\tau = \delta G \Rightarrow \delta = \frac{\tau}{G}$$

$$\tau_\rho = \frac{T\rho}{\bar{J}} \Rightarrow \delta_\rho = \frac{T\rho}{\bar{J}G}$$

$$\Rightarrow d\phi = \frac{T\rho}{\bar{J}G\rho}\,dx \Rightarrow \boxed{\phi = \int_0^l \frac{T}{\bar{J}G}\,dx}$$

$$\Rightarrow \phi = \frac{Tl}{\bar{J}G}$$



$$\phi = \sum_{i=1}^{n} \frac{T_i l_i}{J_i G_i}$$

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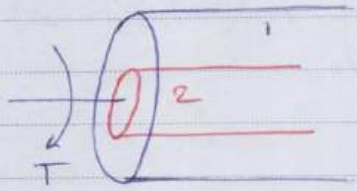
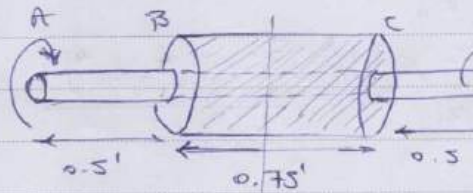
composites



$T = T_1 + T_2$

$\phi_1 = \phi_2$



$800 \times 12 = T_t + T_{sh}$

$\phi_t = \phi_{sh}$

$$\frac{T_t (0.75 \times 12)}{\frac{\pi}{2}(1.5^4 - 1.25^4) G} = \frac{T_{sh}(0.75)(12)}{\frac{\pi}{2}(0.5)^4 G}$$

$$\begin{cases} T_t = 9376.42 \text{ lb-in} \\ T_{sh} = 223.58 \text{ lb-in} \end{cases}$$

$$\phi_{A/D} = \phi_{A/B} + \phi_{B/C} + \phi_{C/D} = \left[\frac{TL}{GJ}\right]$$

$$= 2 \times \frac{800 \times 12 \times 0.5 \times 12}{\frac{\pi}{2}(0.5)^4 (11 \times 10^6)} + \frac{223.58 (0.75) 12}{\frac{\pi}{2}(0.5)^4 (11 \times 10^6)^2}$$

طی ϕ_{BC} هم تا توان چرخش پوسته را در نقطه هم چرخش میله چون فرض

کرده ایم این دو با هم می چرخند.

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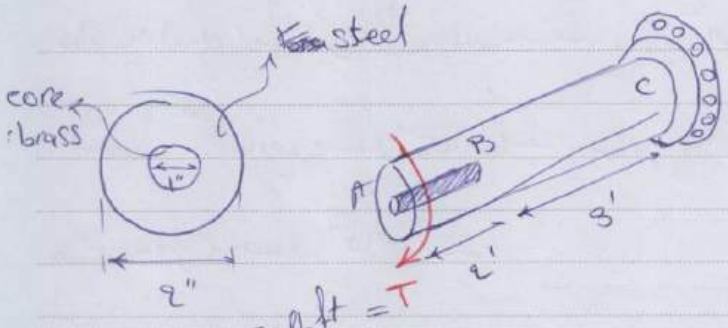
مثال:



$$\tau_{max}|_{brass} = \frac{T_{br}\,0.5}{J_{br}} \qquad T_{br} + T_{st} = 50 \quad (1)$$

$$\phi_{A/B} = \frac{T_{br}(2)(12)}{\frac{\pi}{2}(0.5)^4(5.6)(10^6)} = \frac{T_{st}(2)(12)}{\frac{\pi}{2}(1^4 - 0.5^4)11.5(10^6)} \quad (2)$$

$$\Rightarrow (1),(2) \Rightarrow T_{st} = 48.42$$

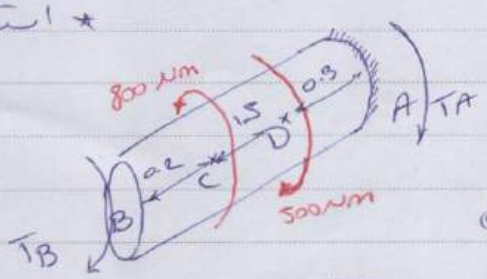
$$T_{br} = 1.57$$

$$\phi = \sum \frac{TL}{JG} = 0.116^\circ$$

τ_{max} برای brasl طبقه رسم آن آنها تا مساحت داخلی steel می‌باشد
در سطح مقطع AB باشد یا BC. AB یا BC یکسان و T یکسان خواهد

* استاتیکی نامعین *

مثال: τ_{max} کدام قسمت؟



$$-T_B + 800 - 500 - T_A = 0$$

$$\phi_{B/A} = 0$$

$$T_A = -345 \text{ Nm}$$

$$T_B = 645 \text{ Nm}$$

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مواد ductile از تنش برشی ضعیف هستند برای این که ماده ductile نیروی کشش فادر شده و ضعیف 45° می‌شکند. مواد brittle از تنش معمولی عمود ضعیف هستند برای این که ضعیف 90° می‌شکند

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$dW = T\,d\theta$ work = torque × angle of twist

$P = \frac{dW}{dt} = T\frac{d\theta}{dt} = T\omega$ (سرعت زاویه‌ای) power = $\frac{work}{time}$

$P = T\omega$ → rad/sec

Watt — Nm

SI: 1 watt = 1 Nm/sec

British: $P = ft.lb/sec$

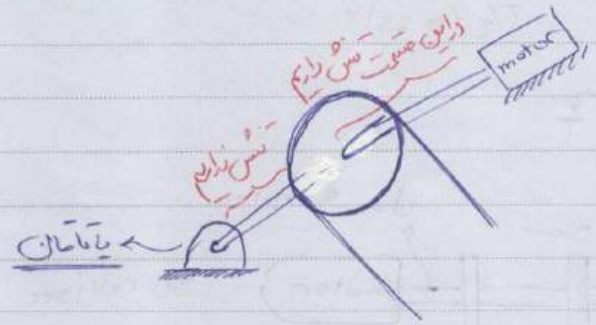
$T = ft.lb$

1 hp = 550 ft.lb/sec

مثال: این شفت توپر برای انتقال قدرت $\omega = 175$ rpm و 5 hp

round per minute

$\tau_{all} = 14.5$ ksi مقاومت خمیری است.



$P = T\omega$ $\omega = 175\,(2\pi)\frac{1}{60} = 18.33$ rad/sec

$P = 5(550) = 2750$ ft lb/sec

$T = \frac{P}{\omega} = 150.1$ ft lb

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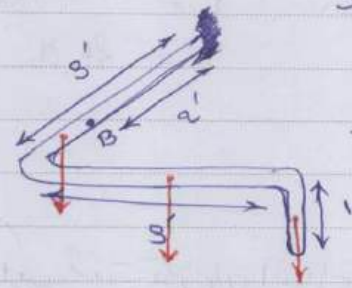
$\frac{\pi}{2}c^4$

$$\tau_{all} = \frac{Tc}{J} \Rightarrow \frac{J}{c} = \frac{T}{\tau_{all}} \Rightarrow c = \left(\frac{2T}{\pi \tau_{all}}\right)^{1/3}$$

$$\Rightarrow c = \left(\frac{2 \times 150.1 \times 12}{\pi (14500)}\right)^{1/3} \Rightarrow c = 0.429'' \Rightarrow d = 0.858''$$

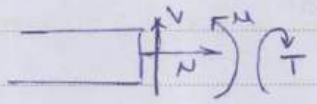
مثال: میله سفر زیر قطر 0.5" وزن 5 lb/ft



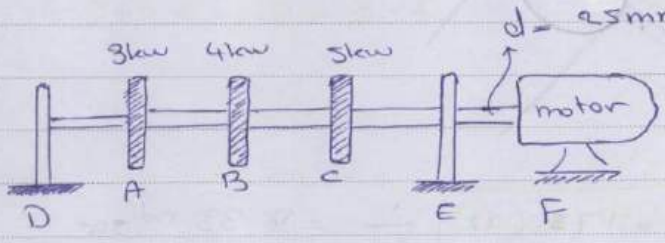
راداست. ماکزیمم تنش برشی را در مقطع B بدست آورید

$$T = \sum M = 15(1.5) + 5(3) = 37.5 \text{ lb.ft} = 450 \text{ lb in}$$

$$\tau_{max} = \frac{Tc}{J} = \frac{450(0.25)}{\pi/2 (0.25)^4} = 18.3 \text{ ksi}$$



مثال:



$$\omega = 50(2\pi) = 100\pi \text{ rad/sec}$$

$$T = \frac{P}{\omega} = \frac{12(10^3)}{100\pi} = 38.2 \text{ Nm}$$

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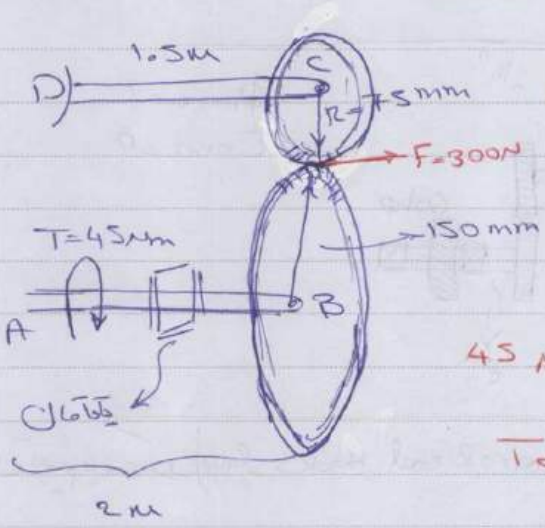
$\tau_{AB} = \frac{3(10^9)}{100\pi} =$

$\tau_{BC} = \frac{7(10^9)}{100\pi} =$

$\tau_{max} = \tau_{CF} = \frac{Tc}{J}$

$\phi_{DF} = \phi = \phi_{AB} + \phi_{BC} + \phi_{CF}$

$\frac{T_{AB}L}{JG}$



$G = 80\ GPa$; Cu

$d = 20\ mm$

Q: twist A = ?

$45\ Nm = F \times 0.15\ m \Rightarrow F = 300N$

$\tau_{CD} = 300 \times 0.075 = 22.5\ Nm$

$\phi_{C/D} = \frac{22.5 \times 1.5}{\frac{\pi}{2}(0.01)^4 (80 \times 10^9)} = 0.0269\ rad$

$\phi_B R_B = \phi_C R_C$

$\phi_B (1.5) = 0.0269 \times 0.75 \Rightarrow \phi_{B/D} = 0.0134\ rad$

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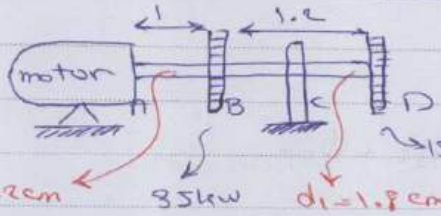
$$\phi_{A/B} = \frac{45 \times 2}{\frac{\pi}{2}(0.01)^4 (80 \times 10^9)} = 0.0716 \text{ rad}$$

$$\phi_{A/D} = \phi_{A/B} + \phi_{B/D} = 0.085 \text{ rad}$$

مثال:



$P = 50 \text{ kw}$

$\omega = 62.8 \text{ rad/sec}$

τ_{all}

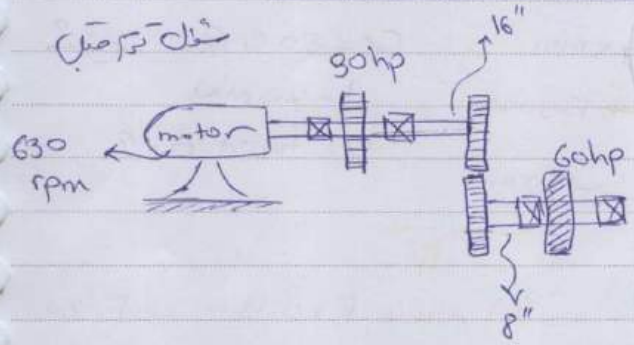
$d = ?$

15kw

$d_2 = 2\text{cm}$ 35kw $d_1 = 1.8\text{cm}$

$35 + 15 \leq 50$ ✓ مرتبه‌ای سوزد!

مثال تمرینی



$\phi_{max} = ?$

$\tau_{max} = ?$

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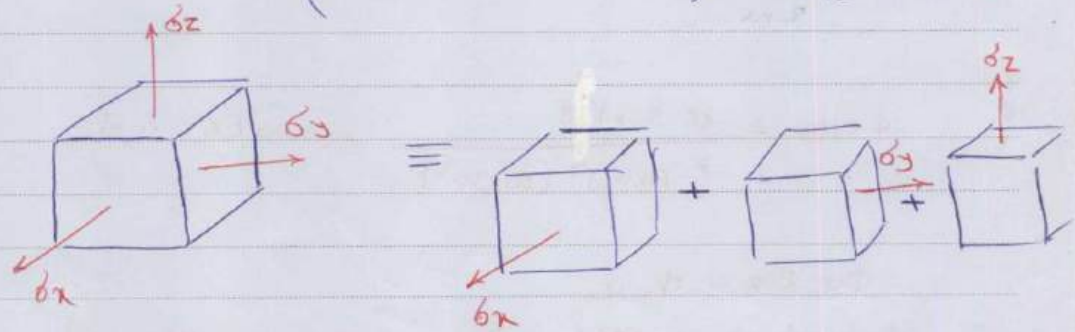
* تانون تعمیم یافته هوک (generalized Hook's law) *



$$\varepsilon_x = \varepsilon'_x + \varepsilon''_x + \varepsilon'''_x = \frac{\sigma_x}{E} - \nu\frac{\sigma_y}{E} - \nu\frac{\sigma_z}{E}$$

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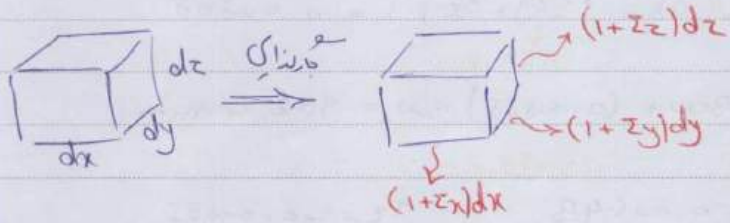
$$\varepsilon_x = \frac{1}{E}\left(\sigma_x - \nu(\sigma_y+\sigma_z)\right)$$

$$\varepsilon_y = \frac{1}{E}\left(\sigma_y - \nu(\sigma_x+\sigma_z)\right)$$

$$\varepsilon_z = \frac{1}{E}\left(\sigma_z - \nu(\sigma_x+\sigma_y)\right)$$



تغییر حجم $\delta V = (1+\varepsilon_x)(1+\varepsilon_y)(1+\varepsilon_z)dxdydz - dxdydz$

$$\Rightarrow \delta V \simeq (\varepsilon_x+\varepsilon_y+\varepsilon_z)\underbrace{dxdydz}_{\Delta V}$$

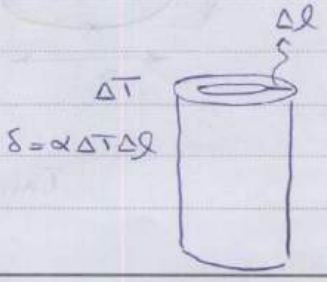
$$e = \frac{\delta V}{\Delta V} = \varepsilon_x+\varepsilon_y+\varepsilon_z$$

کرنش تغییر حجم ایم.

با استفاده از قوانین تعمیم یافته هوک معادله برحسب تنش به صورت زیر می‌توان نوشت:

$$e = \frac{1-2\nu}{E}(\sigma_x+\sigma_y+\sigma_z)$$

$$G = \frac{E}{2(1+\nu)}$$



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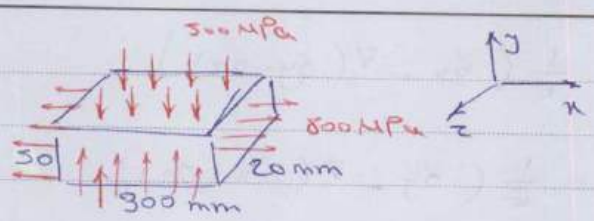
$E = 120\ GPa$

$\nu = 0.34$



: حل

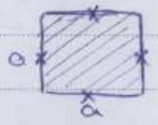
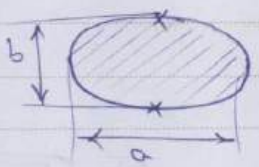
$\sigma_x = 800 MPa$, $\sigma_y = -500 MPa$, $\sigma_z = 0$, $\tau = 0$

$$\varepsilon_x = \frac{1}{E}\left(\sigma_x - \nu(\sigma_y + \sigma_z)\right) = 0.00808$$

طول جدید $= 300 + (0.00808)\,300 = 302.4\ mm$

$\varepsilon_y = -0.00643$ $\varepsilon_z = -0.00085$

تغییر ضخامت $= 19.98\ mm$ تغییر عرض $= 49.68\ mm$

	τ_{max}	ϕ	τ_{max} × زاویه = تغییر طول
	$\frac{4.81T}{a^3}$	$\frac{7.1Tl}{a^4G}$	
	$\frac{20T}{a^3}$	$\frac{46Tl}{a^4G}$	
	$\frac{2T}{\pi ab^2}$	$\frac{(a^2+b^2)Tl}{\pi(a^3b^3)G}$	

$$\tau_{max} = \frac{Tc}{J} \qquad \phi = \frac{Tl}{JG} \quad ; \quad J = \frac{1}{2}\pi c^4$$

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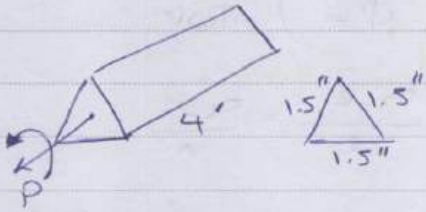
مثال:



$\tau_{all} = 8\,ksi$

$\phi_{All} = 0.02\ rad$

$T_{max} = ?$

$\tau_{all} = \frac{20T}{a^3} \Rightarrow 8\times10^3 = \frac{20T}{(1.5)^3}$

مقایسه با مقطع هم وزن دایروی؟

$\Rightarrow T = 1350\ lb\,in$

$\phi = \frac{46Tl}{a^4 G} = \frac{46T(4\times12)}{(1.5)^4(3.7\times10^6)} \Rightarrow T = 170\ lb\text{-}in$

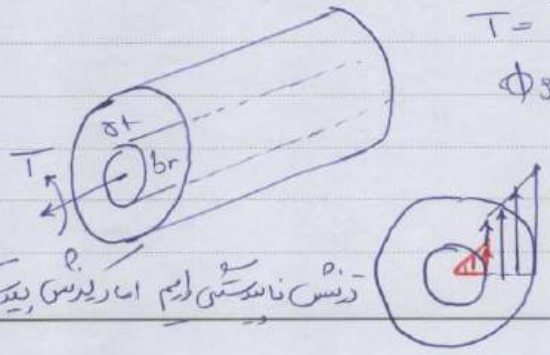
$\Rightarrow T_{max} = 170$

$\rho V_c = \rho V_t \Rightarrow V_c = V_t \Rightarrow \ell A_c = \ell A_t \Rightarrow A_c = A_t$

$\Rightarrow \pi R^2 = \frac{1}{2}(1.5)(1.5)\sin 60 \Rightarrow R = 0.577"$

$\tau_{all} = \frac{TR}{J} \Rightarrow 8\times10^3 = \frac{T(0.577)}{\frac{\pi}{2}(0.577)^4} \Rightarrow T = 2170\ lb\,in$

$\phi_{all} = \frac{Tl}{JG} \Rightarrow T = 233\ lb\,in$



$T = T_{St} + T_{Br}$

$\phi_{St} = \phi_{Br}$ $\Rightarrow$ T_{Br} ✓ T_{St} ✓

تنش نابیشتر داریم اما ریسک بیشتری داریم

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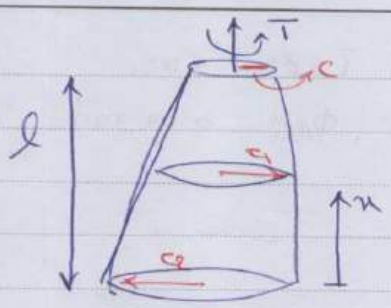

$$\phi = \int_0^{\ell} \frac{T\,dx}{J(x)G}$$

$$\frac{C_2 - C_1}{\ell} = \frac{C_2 - C}{x}$$

$$\Rightarrow C = C_2 - x\,\frac{C_2 - C_1}{\ell}$$

$$J(x) = \frac{\pi}{2}\left(C_2 - x\,\frac{C_2 - C_1}{\ell}\right)^4$$

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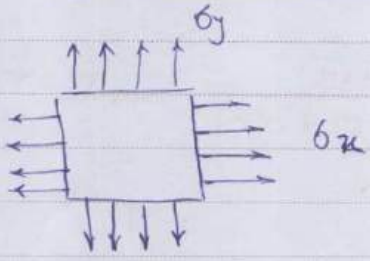

$$\sigma_x = E\left(\frac{\varepsilon_x + \nu\varepsilon_y}{1-\nu^2}\right) \qquad \sigma_y = E\left(\frac{\nu\varepsilon_x + \varepsilon_y}{1-\nu^2}\right)$$

$$\sigma_x = A\varepsilon_x + B\varepsilon_y$$

$$\sigma_y = C\varepsilon_x + D\varepsilon_y$$

$$\sigma_y = 0 \Rightarrow \varepsilon_x = \frac{\sigma_x}{E},\quad \varepsilon_y = -\nu\frac{\sigma_x}{E}$$

$$\sigma_x = 0 \Rightarrow \varepsilon_y = \frac{\sigma_y}{E},\quad \varepsilon_x = -\nu\frac{\sigma_y}{E}$$

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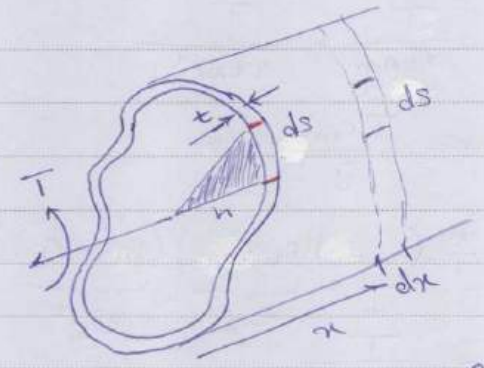
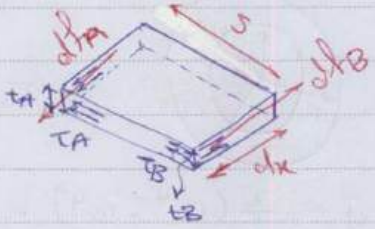




$$\tau_A (t_A dx) = \tau_B (t_B dx)$$

$$\Rightarrow \tau_A t_A = \tau_B t_B$$

جریان برشی $= q = \tau_{av} t = const$

$$df = \tau_{ave}\, dA = \tau_{ave}\,(t ds)$$

$$dT = h(df) = h(\tau_{ave}\, t\, ds) \Rightarrow T = \oint h \underbrace{\tau_{ave}\, t}_{q=const}\, ds$$

$$\Rightarrow T = \tau_{ave}\, t \oint h\, ds = 2\,\tau_{ave}\, t \oint dA_m = 2\,\tau_{ave}\, t\, A_m$$

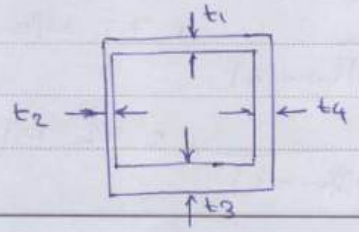
$$dA_m = \tfrac{1}{2} h\, ds$$

$$\Rightarrow \tau_{ave} = \frac{T}{2 t A_m}$$

$$q = \frac{T}{2A_m}$$

D

$$\text{ثابت} \Rightarrow \phi = \frac{TL}{4A_m^2 G} \oint \frac{ds}{t}$$



با ثابت ϕ، دهرگز از 4 قسمت t ثابت،
بین 4 انتگرال جدا گانه می‌گیریم

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$$\tau_{ave} = \frac{T}{2tA_m} = \frac{T}{2t\pi r_m^2}$$

$$\Rightarrow \tau_{ave} = \frac{T r_m}{J}$$

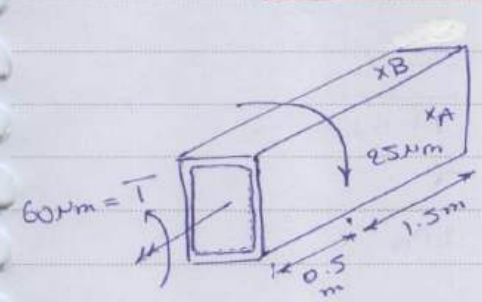
$$J = \frac{\pi}{2}(r_b^4 - r_a^4) = \frac{\pi}{2}\underbrace{(r_b - r_a)}_{t}\underbrace{(r_b + r_a)}_{2r_m}\underbrace{(r_a^2 + r_b)^2}_{2r_m^2} = 2\pi r_m^3 t$$

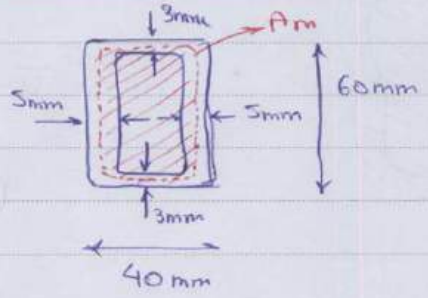
$$\Rightarrow \tau_{ave} = \frac{T r_m}{2\pi r_m^3 t} = \frac{T}{2t\pi r_m^2}$$

$$\phi = \frac{TL}{4A_m^2 G}\oint \frac{ds}{t} = \frac{TL}{4(\pi r_m^2)^2 G t}\oint ds \quad (2\pi r_m)$$

ثابت

$$\Rightarrow \phi = \frac{TL}{\underbrace{(2\pi r_m^3 t)}_{J} G} = \frac{TL}{JG}$$





τ_A, τ_B ? total ϕ

$$\tau_A = \frac{T}{2t_A A_m} \qquad A_m = (0.057)(0.035) = 0.002\ m^2$$

$$\tau_A = \frac{95}{2(0.005)(0.002)} = 1.75\ MPa$$

$$\tau_B = \frac{95}{2(0.003)(0.002)} = 2.92\ MPa$$

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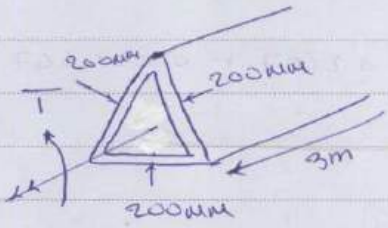
$$\phi = \sum \frac{TL}{4Am^2G} \oint \frac{ds}{t} = \frac{60 \times (0.5)}{4(0.002)^2 \times 35(10^9)} \left[2\frac{57}{5} + 2\frac{35}{3} \right]$$

$$+ \frac{35(1.5)}{4(0.002)^2 \times 35(10^9)} \left[2\frac{57}{5} + 2\frac{35}{3} \right] = 6.29\,(10^{-3})\ \text{rad}$$



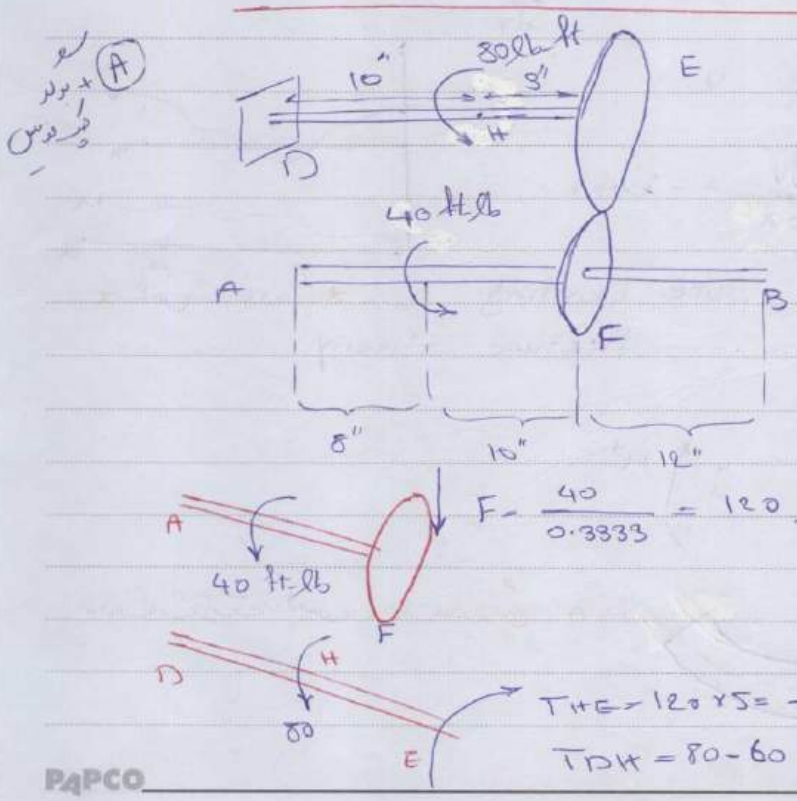
$\tau_{all} = 90\ MPa$

made of 5mm plate

$\phi_{max} = 2 \times 10^{-3}$ rad

$T_{max} = ?$

$T_1 = 15.6\ kN.m$, $T_2 = 500\ NM$ } $T_{max} = 500\ N.m$



$\phi_{A)\ total} = ?$

$$F = \frac{40}{0.3333} = 120\ lb$$

$T_{HE} = 120 \times 5 = -60$

$T_{DH} = 80 - 60 = 20$

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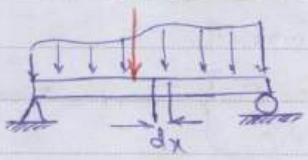
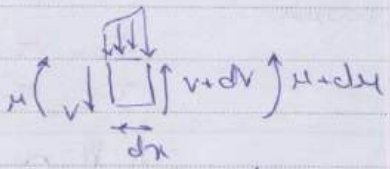
$$\Phi_E = \frac{-60(12\times30) + 20(12\times10)}{\frac{\pi}{2}(0.5)^4(11\times10^6)} = 0.01778 \text{ rad "cw"}$$

$$\Phi_F = \frac{6}{4}\Phi_E = 0.02667 \text{ (ccw)}$$

$$\Phi_{A/F} = \frac{-40(12)(10)}{\frac{\pi}{2}(0.5)^4(11\times10^6)} = -0.004445 \text{ (ccw)}$$

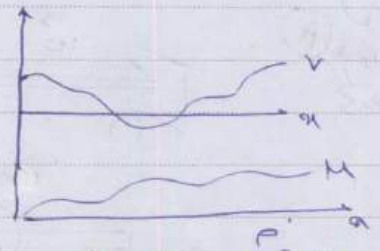
$$\Phi_{A/total} = \Phi_F + \Phi_{A/F} = 0.02667 + 0.004447 = \text{---}$$





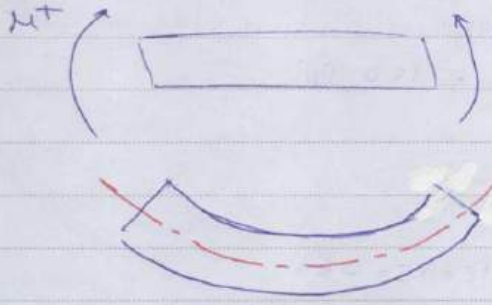
معادلات تعادل برای المان

$$\begin{cases} \frac{dM}{dx} = V \\ \frac{dV}{dx} = -w(x) \end{cases}$$



pure bending ★ خمش خالص ★

flexture theory



تغییر شکل ایجاد شده در اثر خمش ⇐ محور خنثی

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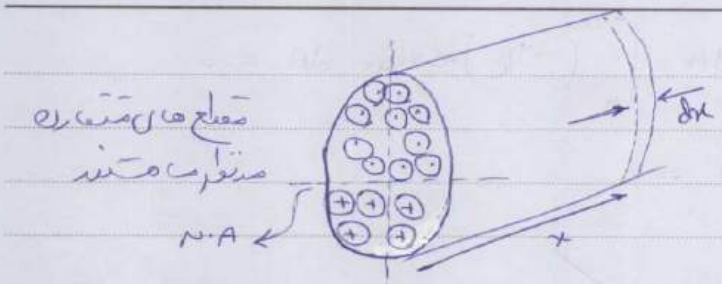
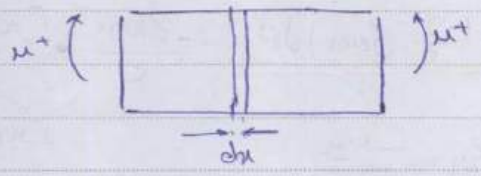
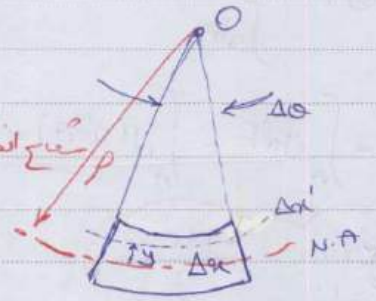




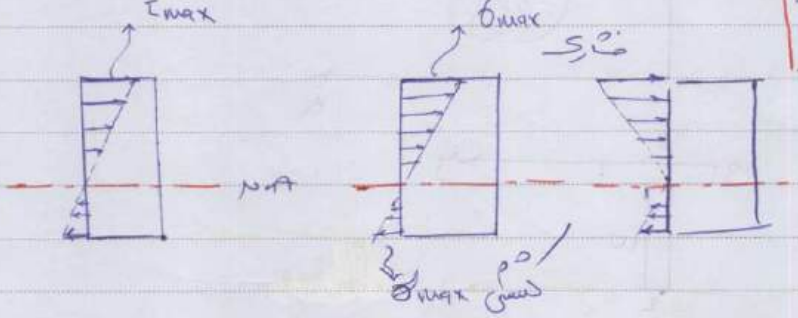


$$\tau_y = \lim_{\Delta x \to 0} \frac{\Delta x' - \Delta x}{\Delta x}$$

$$\Delta x = \rho \Delta\theta \quad , \quad \Delta x' = (\rho - y)\Delta\theta$$

$$\tau_y = \lim_{\Delta x \to 0} \frac{(\rho - y)\Delta\theta - \rho\Delta\theta}{\rho\Delta\theta} \longrightarrow \boxed{\tau_y = \frac{-y}{\rho}}$$

$$\boxed{\tau_y = y/c\ \tau_{max}}$$



$$\sigma = E\varepsilon$$

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$$\sum F = 0 \longrightarrow \int_A \sigma\, dA = \int_A (-y_k)\,\sigma_{max}\, dA = 0$$

$$\Longrightarrow \frac{\sigma_{max}}{c}\int_A y\, dA = 0$$

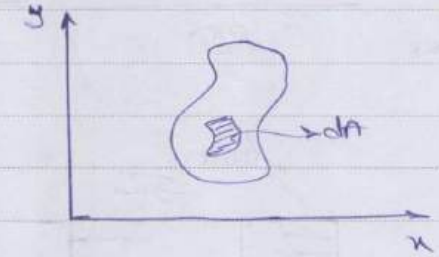
$$\bar{y} = \frac{\int y\, dA}{\int dA}$$ حداقل فاصله از مرکز سطح (centroid) سطح مقطع عرضی تیر

$$M = \int_A y\, dF = \int_A y(\sigma\, dA) = \int_A y\left(\frac{-y}{c}\sigma_{max}\right) dA = \frac{-\sigma_{max}}{c}\underbrace{\int y^2 dA}_{I_{xx}}$$

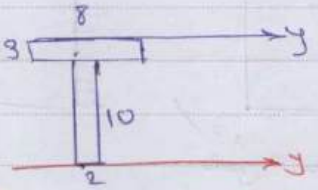
$\sigma_{max} = \frac{-Mc}{I}$ در تیر با مقطع متقارن

$$\sigma_y = \frac{-My}{I}$$

$$\bar{y} = \frac{\int y\, dA}{\int dA} = \frac{\sum \bar{y}A}{\sum A}, \quad I_{xx} = \int y^2 dA$$



مثال:



$$\bar{y} = \frac{\sum \bar{y}A}{\sum A} = \frac{(2\times10)(5) + (3\times8)(11.5)}{2\times10 + 3\times8} = 8.55$$ از بالا

$$\bar{y} = \frac{(3\times8)(-1.5) + (10\times2)(-8)}{3\times8 + 10\times2} = -4.45$$ از پایین

PqPCO

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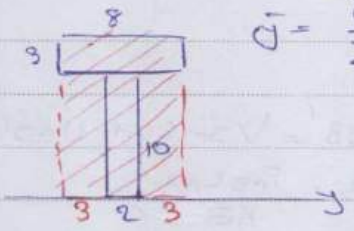
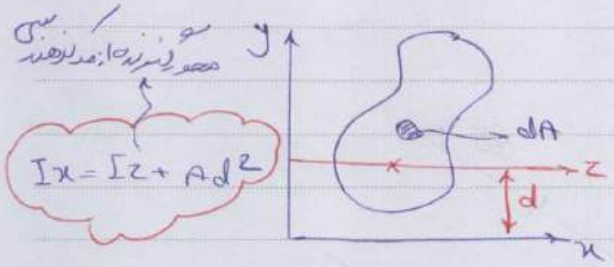


$$\bar{y} = \frac{\Sigma \bar{y} A}{\Sigma A} = \frac{(13\times8)(6.5) - 2(10\times3)(5)}{13\times8 - 2\times(10\times3)}$$

$$= 8.55$$

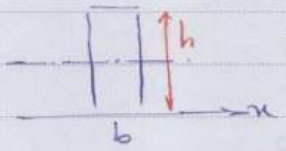


$I_x = I_z + Ad^2$ (سطح دیسک را بدهید)

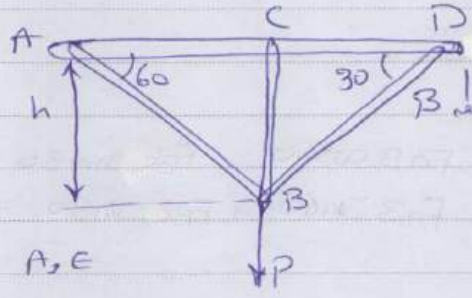
$$I_x = \int_A y^2 dA$$

$$I_y = \int_A x^2 dA$$

$$J = \int_A r^2 dA = \int_A (x^2+y^2)\, dA = I_x + I_y$$



$$I_{N.A} = \tfrac{1}{12} b h^3 \Rightarrow I_x = \tfrac{1}{12} b h^3 + (bh)(h/2)^2 = \tfrac{1}{3} b h^3$$



جابجایی افقی و عمودی نقطه B

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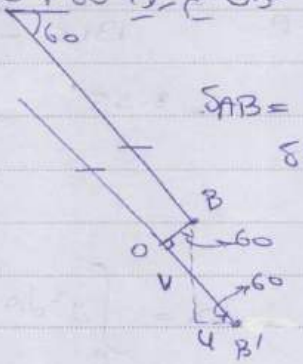
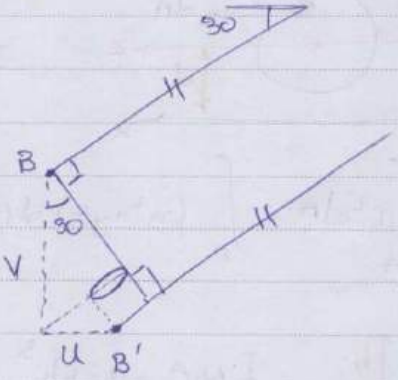
نوشتن تعیین زاویه 60 و 60 بقیه کانه



$$\delta_{AB} = OB' = V\sin 60 + U\cos 60$$

$$\delta_{AB} = \frac{F_{AB} L_{AB}}{AE}$$



$$\delta_{BD} = V\sin 30 - U\cos 30 = \frac{F_{BD} L_{BD}}{AE}$$

$$\delta_{BC} = V = \frac{F_{BC} L_{BC}}{AE}$$

$$\sum F_x = 0 \Rightarrow F_{AB}\cos 60 = F_{BD}\sin 30$$

$$\sum F_y = 0 \Rightarrow F_{AB}\sin 60 + F_{BD}\sin 30 + F_{BC} = P$$

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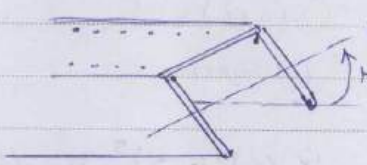
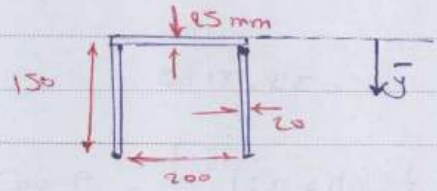
مثال: تنش‌های وارد بر نقاط فوقانی و تحتانی را محاسبه کنید





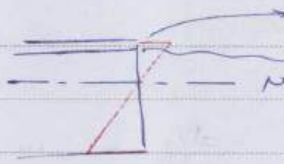
$$\bar{y} = \frac{\sum \bar{y}A}{\sum A} = \frac{(0.24 \times 0.025)(0.0125) + 2(0.15 \times 0.02)(0.025 + 0.075)}{(0.24 \times 0.025) + 2(0.15 \times 0.02)}$$

$$= 0.05625\ m$$

$$I = \sum \left(\frac{bh^3}{12} + Ad^2 \right) = \frac{1}{12}(0.24)(0.025)^3 + (0.24)(0.025)(0.05625 - 0.0125)^2$$

$$+ 2 \times \frac{1}{12}(0.02)(0.15)^3 + 2(0.15)(0.02)(0.04375)^2$$

$$= 34.53 \times 10^{-6}\ m^4$$



$$\sigma_{max} = \frac{600(0.05625)}{34.53 \times 10^{-6}} = 0.977\ MPa$$

$$\sigma_{min} = \frac{600(0.175 - 0.05625)}{34.53 \times 10^{-6}} = 0.593$$

$$F = \frac{(0.977 + 0.593)}{2}(0.025 \times 0.24) = 4.56\ kN$$

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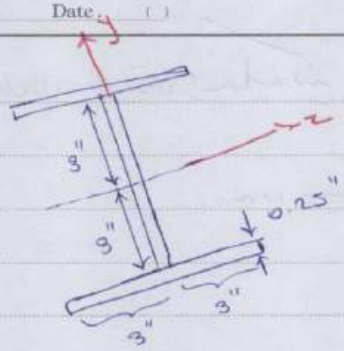


مثال

$\sigma_{all} = 24$ ksi

$(M_{all})_z$

$(M_{all})_y$

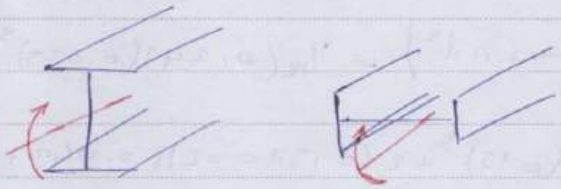
$$I_z = \tfrac{1}{2}(6)(6.5)^3 - \tfrac{1}{12}(6-0.25)6^3$$

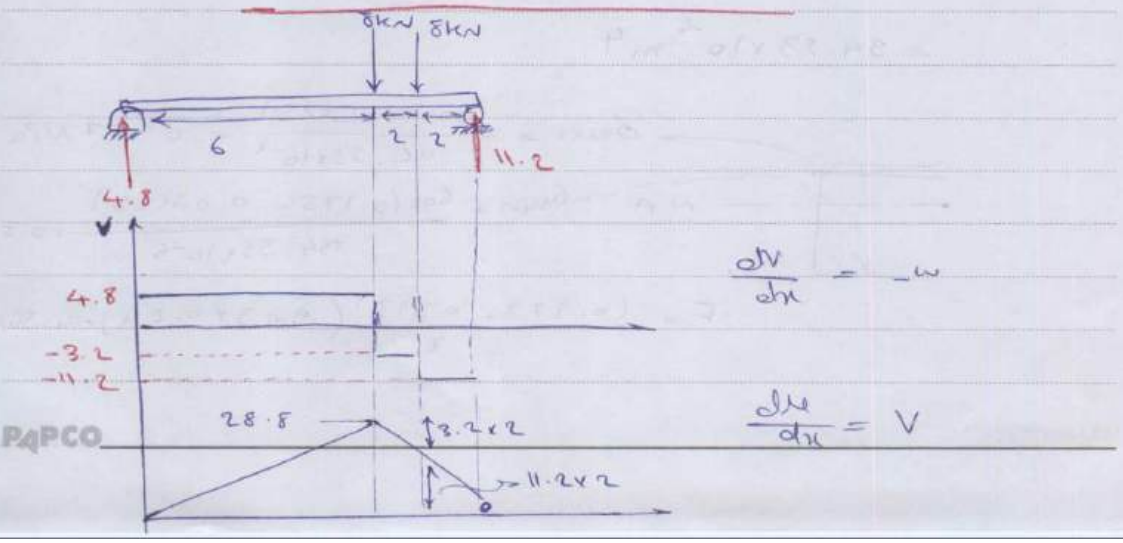
$$= 33.8125$$

$$I_y = 2\left(\tfrac{1}{12}(0.25)(6)^3 + \tfrac{1}{12}(6)(0.25)^3\right) = 9.0078$$

$$(M_{all})_z = \frac{\sigma_{all} I_z}{c} = \frac{24(33.8125)}{3.25} = 249.7 \text{ kip-in}$$

$$(M_{all})_y = \frac{\sigma_{all} I_y}{c} = \frac{24(9.0078)}{3} = 72.0625 \text{ kip-in}$$





$$\frac{dV}{dx} = -w$$

$$\frac{dM}{dx} = V$$

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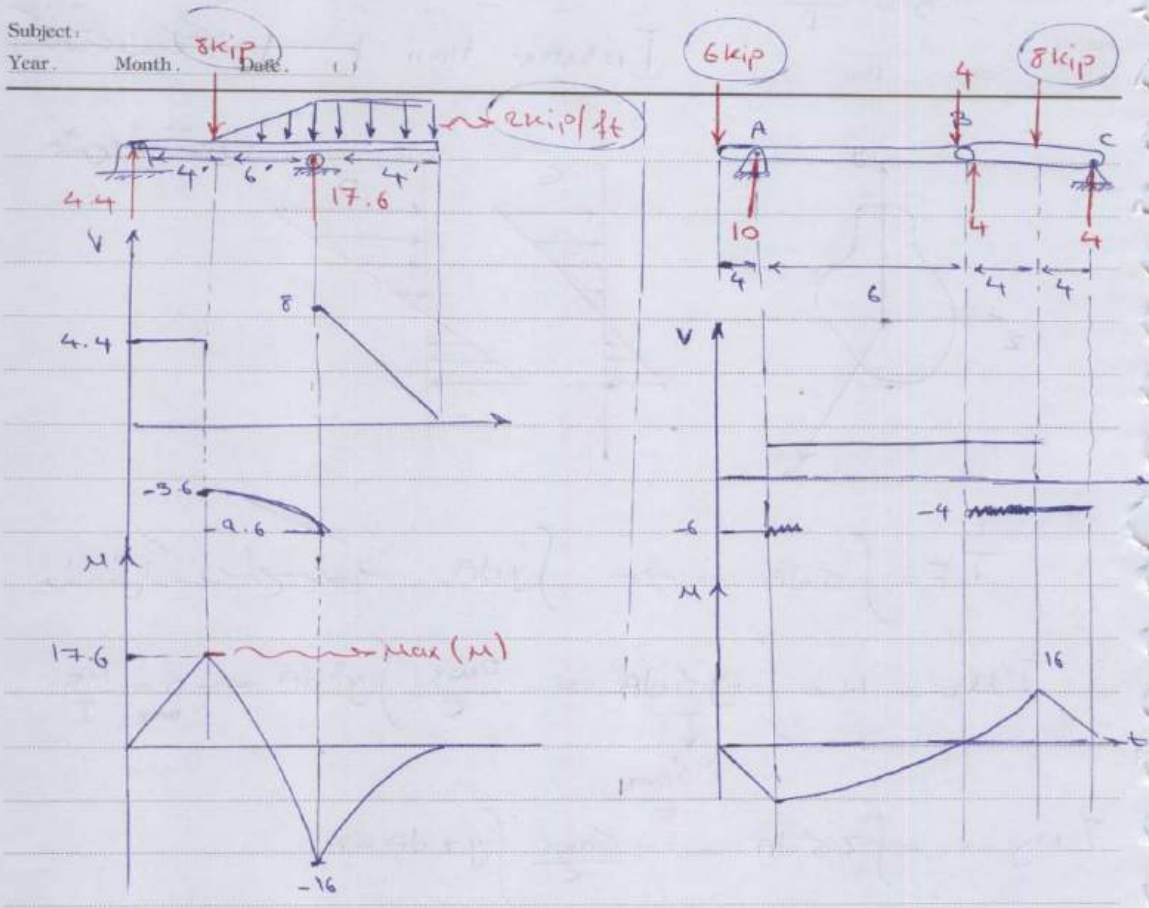

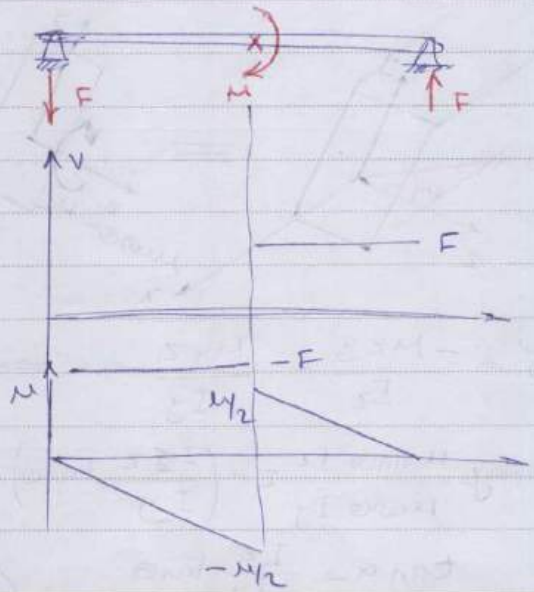

PqPCO

$\sigma = \frac{My}{I}$ ← فرمول تنش خمشی را به این شکل می‌نویسیم که I نشان دهنده شیب تیر را در آن مقطع بدست آورد

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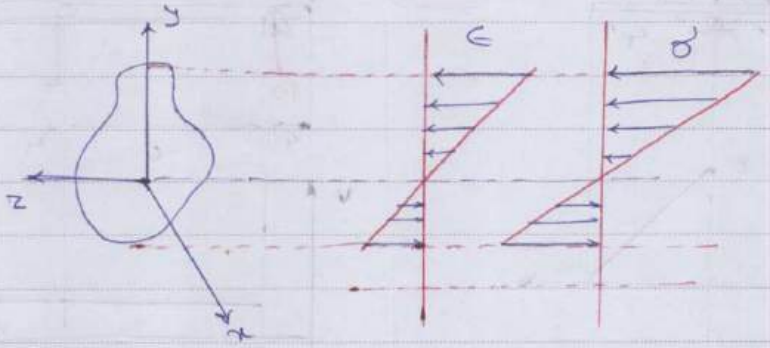
I is better than ⊢ که بهتر وارد شود.

تنش غیر متقارن



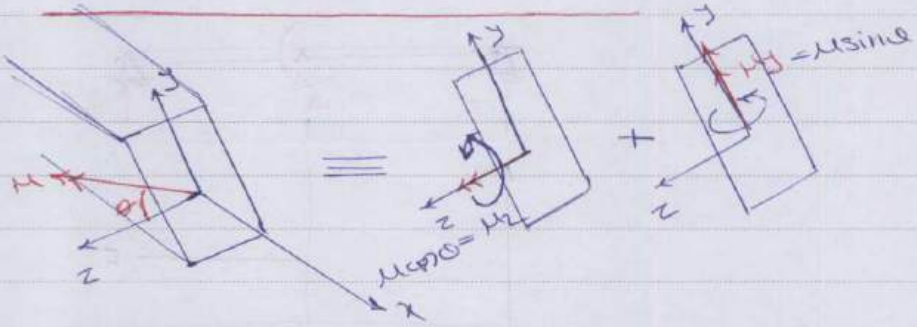
$\sum F = \int \sigma \, dA = 0 \Rightarrow \int y \, dA$ خنثی از مرکز سطح عبور می‌کند

$\sum M_z = M = \int -y\sigma \, dA = \frac{\sigma_{max}}{c} \int y^2 dA \Rightarrow \sigma_{max} = \frac{Mc}{I}$

$\frac{\sigma \sigma_{max}}{c}$

$\sum M_y = 0 = \int z\sigma \, dA = -\frac{\sigma_{max}}{c} \int yz \, dA = 0$

مثال:



$\sigma = \frac{-M_z y}{I_z} + \frac{M_y z}{I_y} = 0 \Rightarrow y = \frac{M_y I_z}{M_z I_y} z$

$\Rightarrow y = \frac{M \sin\theta \, I_z}{M \cos\theta \, I_y} z = \left(\frac{I_z}{I_y} \tan\theta\right) z$

$\tan\alpha = \frac{I_z}{I_y} \tan\theta$ شیب خنثی از مرکز سطح

PdPCO

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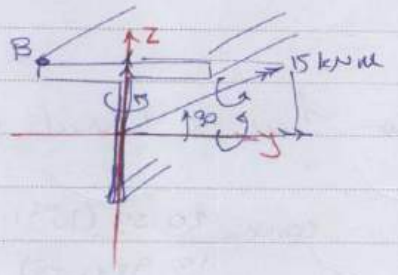
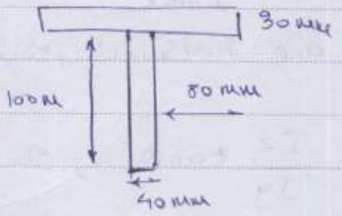
حل:





$M_y = 15\cos 30 = 12.99\ kNm$

$M_z = 15\sin 30 = 7.5\ kNm$

$$\bar{z} = \frac{\sum \bar{z}A}{\sum A} = \frac{0.05(0.1)(0.04) + 0.115(0.08)(0.2)}{(0.1)(0.04) + (0.08)(0.2)}$$

$\bar{z} = 0.089\ m$ از پایین

$$I_z = \frac{1}{12}(0.1)(0.04)^3 + \frac{1}{12}(0.08)(0.1)^3 = 20.53(10^{-6})\ m^4$$

$$I_y = \left[\frac{1}{12}(0.04)(0.1)^3 + (0.1)(0.04)(0.089 - 0.05)^2\right] + \left[\frac{1}{12}(0.2)(0.08)^3 + (0.2)(0.08)(0.115 - 0.089)^2\right]$$

$$= 19.92(10^{-6})\ m^4$$

$$\sigma_B = \frac{-M_z y}{I_z} + \frac{M_y z}{I_y} = -\frac{(7.5)(-0.1)}{20.53\times10^{-6}} + \frac{12.99(0.041)}{19.92\times10^{-6}} = 74.8\ MPa$$

$$\sigma_C = \frac{-7.5(0.02)}{20.53(10^{-6})} + \frac{12.99(-0.089)}{19.92(10^{-6})} = -90.3\ MPa$$

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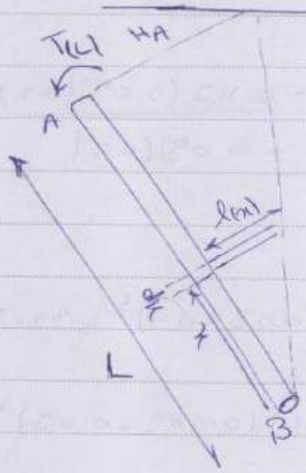
y represents I_{min}

z " I_{max}

θ & α are measured positive from +z towards +y axis.

$$\tan\alpha = \frac{I_z}{I_y}\tan\theta \;;\; \theta = 60 \qquad \tan\alpha = \frac{20.33\,(10^{-6})}{13.92\,(10^{-6})}\tan 60$$

$$\Rightarrow \alpha = 68.6$$



9-4-14

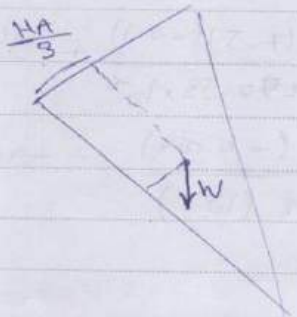
$$T(x) = \underbrace{\rho g}_{k}\, l(x)\left(\frac{l(x)}{2}\right)dx$$

$$l(x) = \frac{H_A x}{L}$$

$$W = \int_0^L k\,l(x)\,dx \Rightarrow k = \frac{2W}{LH_A}$$

$$T(L) = \int_0^L \frac{k H_A^2}{2L^2}\,x^2\,dx \Rightarrow T(L) = \frac{H_A W}{3}$$

$$\Rightarrow T_{max} = \dots$$



P4PCO

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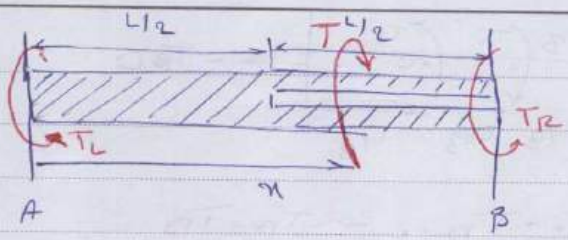


9-8-7

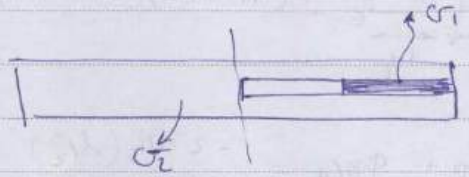
$T_R + T_L = T \Rightarrow T_R = T_L \Rightarrow T_R = T_L = T/2$

$$\phi_R = \frac{T_R (L-x)}{G J_H}$$

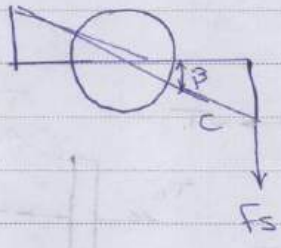
$$\phi_L = \frac{T_L (x - L/2)}{G J_H} + \frac{T_L \, L/2}{G J_N}$$

$\rightarrow \phi_R = \phi_L$

$$\Rightarrow \frac{T/2\,(L-x)}{J_H} - \frac{T(x-L/2)}{J_H} = \frac{\frac{1}{4}L}{J_N}$$



9-8-8



$F_S = k(b - c\beta) \rightarrow T_S = qk(b - \beta c)c$

$$\beta = \frac{T_S L}{G I_P}$$

$\Rightarrow$ B✓

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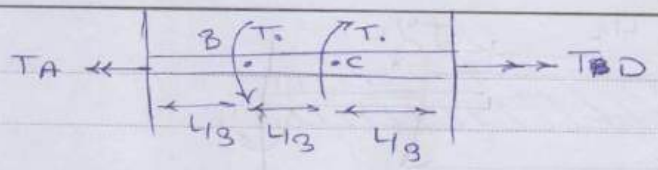


3-22

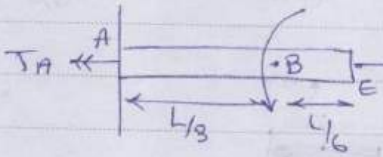
$$-T_A + T_0 - T_0 + T_D = 0 \Rightarrow T_A = T_D$$

$$\phi_{AD} = 0 \Rightarrow \phi_{A/B} + \phi_{B/C} + \phi_{C/D} = 0$$

$$\Rightarrow \frac{T_A(l/3)}{JG} + \frac{(T_A - T_0)\,l/3}{JG} + \frac{(T_D)\,l/3}{JG} = 0$$

$$\Rightarrow T_A = T_D = T_0/3$$

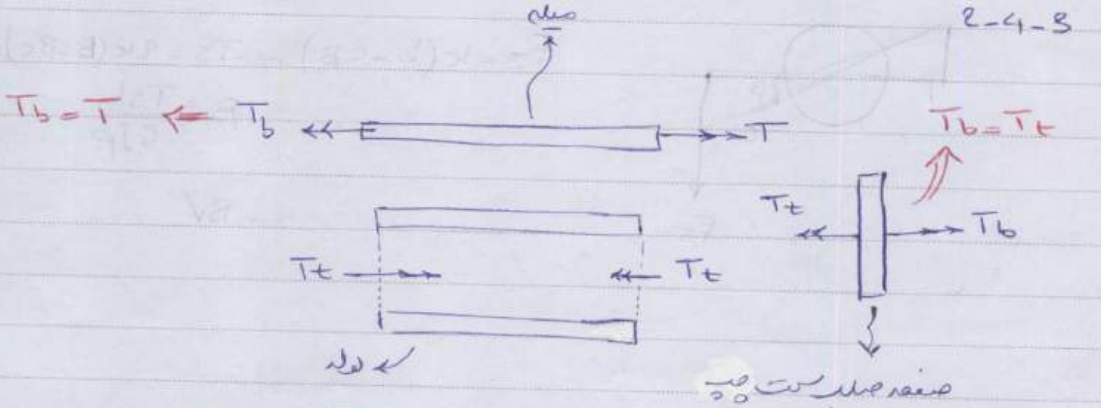
$$\phi_{max} = \phi_{AB} = \frac{\frac{T_0}{3}(l/3)}{JG}$$



$$(T_A - T_0) = -2T_0/3$$

$$\phi_{E/A} = \phi_{E/B} + \phi_{B/A} = \frac{-2T_0/3\,(l/6)}{JG} + \frac{T_0/3\,(l/3)}{JG} = 0$$

2-4-3



$$\tau_{max} = \frac{Tc}{J}$$

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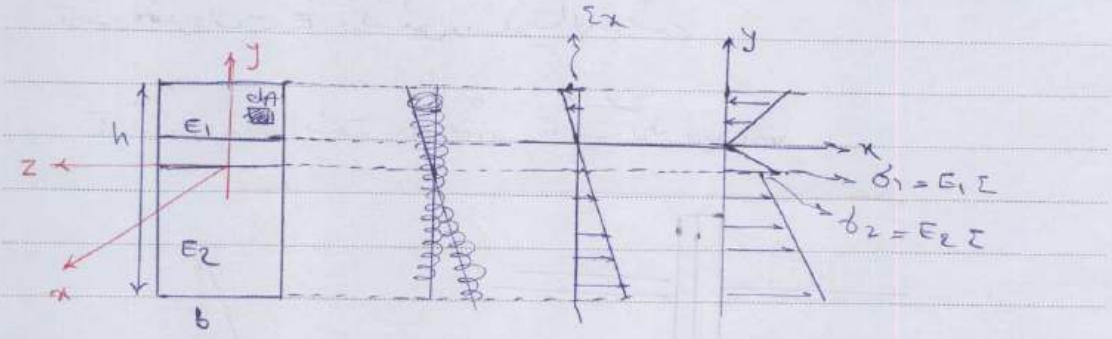


$E_1 > E_2$

$dA = dy\,dz$

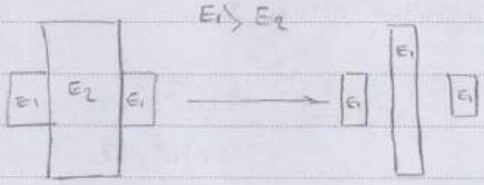
$\underbrace{\sigma_1 dA_1}_{dF_1} = (E_1 \varepsilon)\, dy\, dz = dF_2 = \sigma_2 dA_2 = (E_2 \varepsilon) n\, dy\, dz$

$\Rightarrow n = \frac{E_1}{E_2}$

$y = const \quad dy = const \quad dz_2 = n\, dz_1$

$E_1 > E_2$

$E_1 > E_2$



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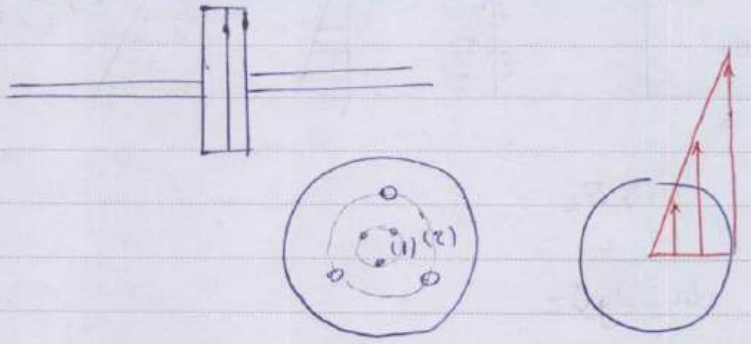
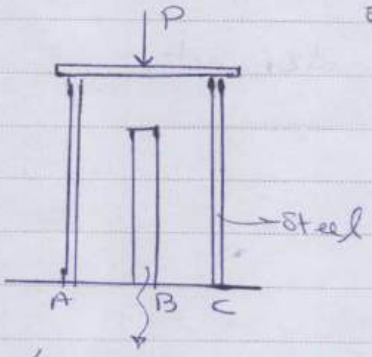
توجه: ممکن است E در کشش و فشار یکسان نباشد

I اشکال خاص نیست به جز حین ... + پیچ بدون فرض ها



$$T = (\text{تعداد پیچ}) * \left(\frac{r_1}{r_2} T_{all} * A * \text{شعاع}\right) + (\text{تعداد پیچ}) * (T_{all} * A * \text{شعاع})$$

(1) (2)



$E_S = 8\times10^6$ psi

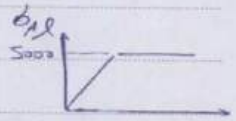
$l = 10"$, $A = 2in^2$

$E_{Al} = 10^7$ psi

$\sigma_y = 5000$ psi

$l = 9.995"$

$A = 4in^2$



الاستیک پلاستیک کامل

(a) gap = 0.005"

(a) P = 30 kips

(b) P = 200 kips

$$\Delta l_S = \frac{15\times10^3 \times 10}{8\times10^6 \times 2} = 0.025" > 0.005"$$

P ... gap ... سفت می‌شود ... سیستم ... صلب می‌شود

PAPCO

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$$\begin{cases} 2F_S + F_{Al} = P \quad (1) \\ \Delta_S = \Delta_{Al} + gap \quad (2) \end{cases} \Rightarrow \frac{F_S(10)}{3\times10^6 \times 2} = \frac{F_{Al}(10)}{4\times10^7} + 0.005$$

$9.995 \simeq$

$(1),(2) \Rightarrow F_S = 5769.23$

$F_{Al} = 18462.54 \qquad \sigma_{Al} = \frac{F_{Al}}{4} = 4615.69 < \sigma_y$

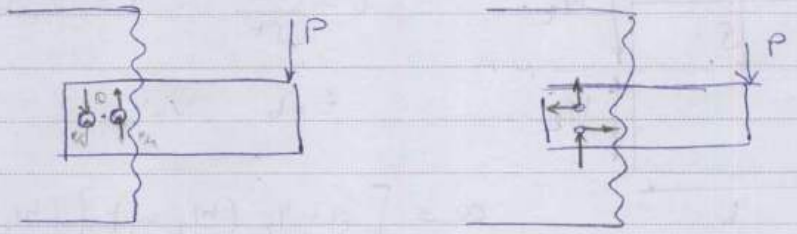
(b) $F_S = 25984.6 \qquad F_{Al} = 19928.78$

$\sigma_{Al} = 9730.7.94 > \sigma_y \rightarrow$ not accepted

چون آلومینیم بیشتر از σ_y تسن تحمل نمی‌کند

$\sigma_{Al})_{max} = 5000 \Rightarrow F_{Al})_{max}) = 5000 \times 4 = 20000\ lb = R_B$

$F_S = \frac{200000 - 20000}{2} = 90000 = R_A = R_C$



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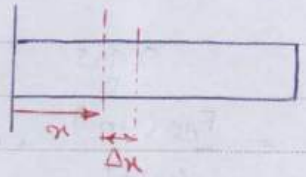
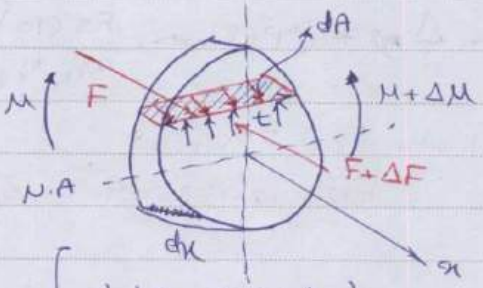
* فصل ۵ * ممتنات



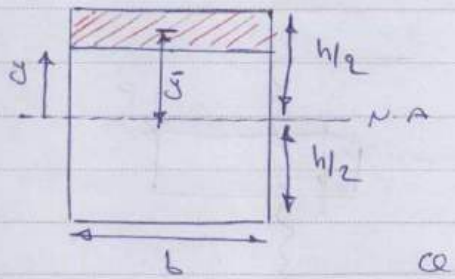


$$\sum F_x = 0 \Rightarrow \int_{A'} \sigma' dA - \int_{A'} \sigma dA - \tau (t\,dx) = 0$$

$$\Rightarrow \int_{A'} \left(\frac{M + dM}{I}\right) y\,dA - \int_{A'} \frac{M}{I} y\,dA - \tau t\,dx = 0$$

$$\Rightarrow \frac{dM}{I} \int_{A'} y\,dA = \tau t\,dx \Rightarrow \tau = \frac{1}{It}\left(\frac{dM}{dx}\right)\int_{A'} y\,dA$$

$$\Rightarrow \tau = \frac{VQ}{It} \; ; \; Q = \int_{A'} y\,dA \; ; \; V = \frac{dM}{dx}$$



$$\tau = \frac{VQ}{It} \qquad I = \frac{1}{12} b h^3$$

$$t = b \qquad V = V$$

$$Q = \underbrace{\left[y + \frac{1}{2}\left(\frac{h}{2} - y\right) \right]}_{\bar{y}} \underbrace{\left(\frac{h}{2} - y\right) b}_{A}$$

$$\Rightarrow Q = \frac{1}{2}\left(\frac{h^2}{4} - y^2\right) b$$

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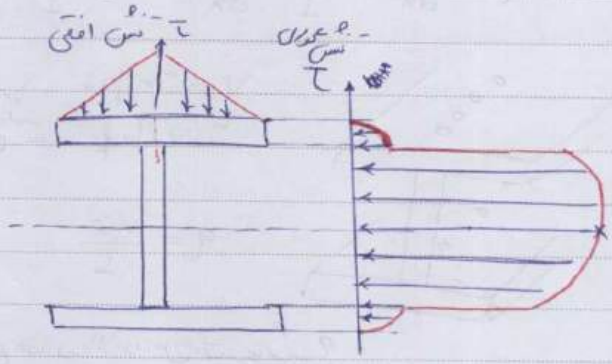
$$\tau = \frac{V\,Q/I\,(\frac{h^2}{4}-y^2)b}{\frac{1}{12}bh^3 b} = \frac{6V}{bh^3}\left(\frac{h^2}{4}-y^2\right)$$

$y = h/2 \Longrightarrow \tau = 0$ $\quad y = 0 \Longrightarrow \tau_{max} = \frac{6V}{4bh} = 1.5\frac{V}{A}$

$$\tau = \frac{6V}{bh^3}\left(\frac{h^2}{4}-y^2\right)$$

$$\int_{-h/2}^{h/2} \frac{6V}{bh^3}\left(\frac{h^2}{4}-y^2\right)\underbrace{b\,dy}_{dA} = \frac{6V}{h^3}\left[\frac{h^2}{4}y - \frac{y^3}{3}\Big|_{-h/2}^{h/2}\right] = V$$



* تنش برشی مانند تنش عمودی در سطح خنثی آن مقدار صفر است *

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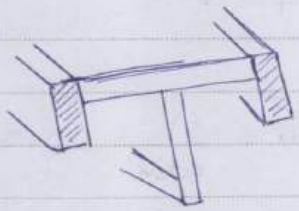
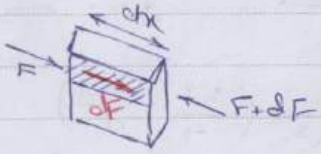
جریان برشی: نیروی برشی به ازای واحد طول $q = \frac{dF}{dx}$



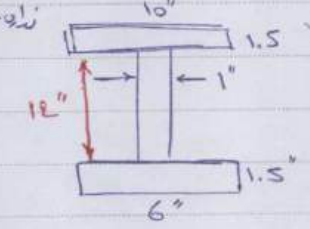


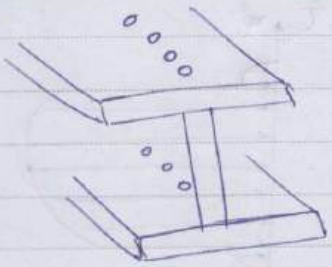
$$q = \frac{VQ}{I}$$

$$dF = \sigma \int_A dA = \frac{dM}{I} \int y\,dA = \frac{dM}{I} Q$$

$$q = \frac{dF}{dx} = \frac{1}{I} \frac{dM}{dx} Q = \frac{VQ}{I}$$

مثال: فاصله جوش‌های بال تیر (مجموع نیروهای برشی) را بیابید.





مثال: $V = 5 kips$

نیروی برشی در هر متر طول جوش $500 lb/ft$

$$q = \frac{VQ}{I}$$

فاصله جوش‌ها در بالا و پایین چقدر باشد؟

$$\bar{y} = \frac{\sum \bar{y} A}{\sum A} = 8.625 in$$

$$I_{N.A} = 1196.4\ in^4$$

$$q_t = \frac{500 (84.375)}{1196.4} = 352.6\ lb/in \Rightarrow S_t = \frac{500}{352.6} = 1.42$$

$$q_b = \frac{500 (70.87)}{1196.4} = 296.19\ lb/in \Rightarrow S_b = \frac{500}{296.19} = 1.69$$

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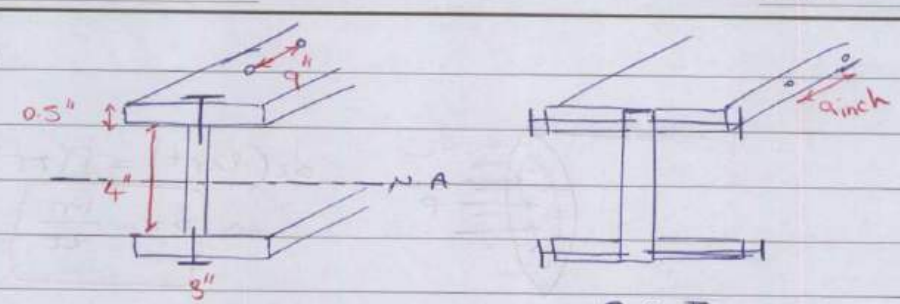
مثال:



ماکزیمم بتواند "V" را در صورت بیت آورید. قدرت هر میخ = 40 lbf

$$I = \frac{1}{12} \times 3 \times 5^3 - 2 \times \frac{1}{12} \times 1 \times 4^3 = 20.58 \text{ in}^4$$

case I: $q = \frac{VQ}{I} \Rightarrow \frac{40}{9} = \frac{V(3 \times 0.5) \times (2.25)}{20.58} \Rightarrow V = 27.1 \text{ lb}$

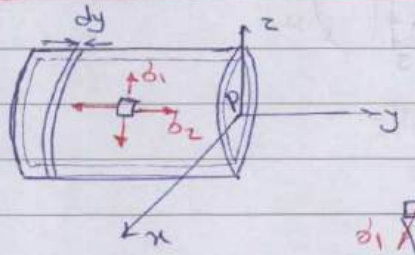
case II: $q = \frac{VQ}{I} \Rightarrow \frac{40}{9} = \frac{V(1 \times 0.5) \times (2.25)}{20.58}$

مخازن جدارنازک:

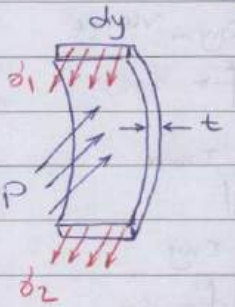
σ_2 = تنش طولی

σ_1 = تنش محیطی

تنش "hoop"



$\frac{r}{t} \geq 10$



$$\Sigma F_x = 0 \Rightarrow 2(\sigma_1)(t\,dy) - P(2r\,dy) \quad P$$

$$\Rightarrow \boxed{\sigma_1 = \frac{Pr}{t}}$$

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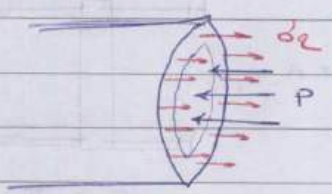
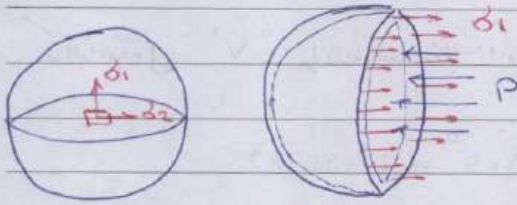


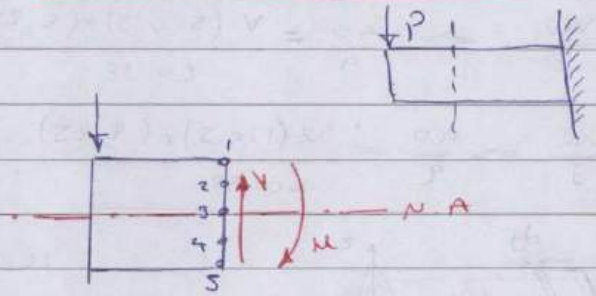
$$\sigma_2(2\pi rt) = P(\pi r^2)$$

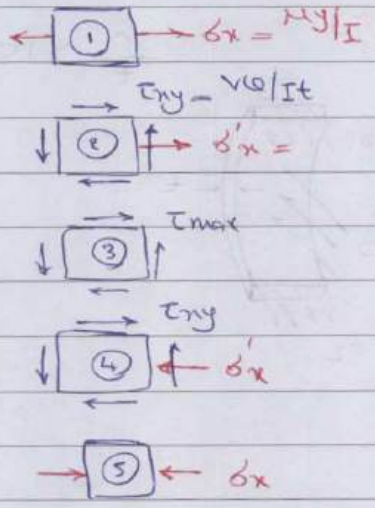
$$\Rightarrow \boxed{\sigma_2 = \frac{Pr}{2t}}$$



$$\sigma_1(2\pi rt) = P(\pi r^2)$$

$$\Rightarrow \boxed{\sigma_1 = \sigma_2 = \frac{Pr}{2t}}$$





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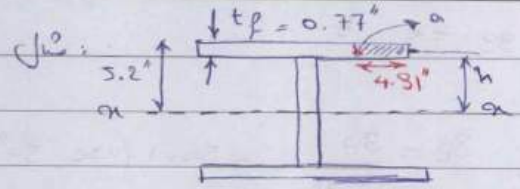


$I_{xx} = 994\ in^4$

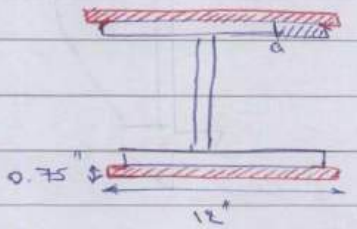
$V = 50\ kips$

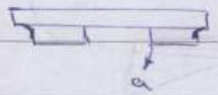
$\tau_a = \frac{VQ}{It}$

$h = 5.2 - \frac{0.77}{2} = 4.815"$

$Q = (4.31 \times 0.77) \times 4.815 = 19.98\ in^3$

$\tau = \frac{50}{394} \times \frac{19.98}{0.77} = 2.63\ ksi$





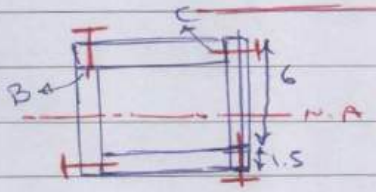
$Q = 2\left(4.31 \times 0.77 \times 4.815\right) + \left(12 \times 0.75 \times 5.575\right)$

$= 82.1\ in^3$

$I = 994 + 2\left(\frac{1}{12} \times 12 \times (0.75)^3 + (12 \times 0.75)(5.575)^2\right) = 954\ in^4$

$\tau_a = \frac{50 \times 82.1}{954 \times (1.54)} = 2.79\ ksi$

2×0.77



وزنه: 80 lb عزم مقاومت = ؟

$I_{N.A} = \frac{1}{12}(7.5)^4 - \frac{1}{12}(4.5)^4 = 229.5\ in^4$

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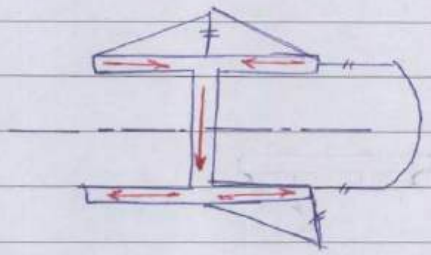
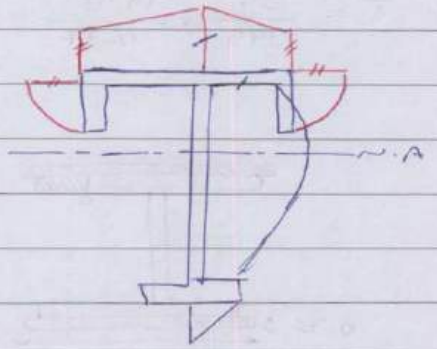
$Q_B = \left(\frac{4.5}{2} + 0.75\right)(1.5 \times 7.5) = 93.75\ in^3$

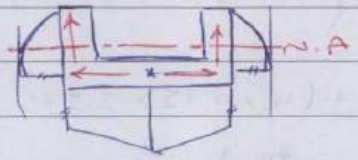
$q_B = \frac{VQ}{I} = 11.76\ lb/in \qquad S_B = \frac{90}{\frac{11.76}{2}} = 50.1"\ (use\ 50")$

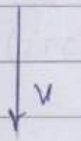
$q_C = \frac{VQ}{I} = \frac{80 \times 20.25}{229.5} = 7.059\ lb/in$

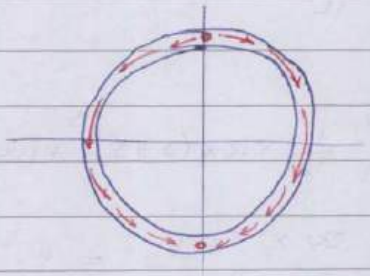
$Q_C = 3(4.5 \times 1.5) = 20.25 \qquad S_C = \frac{90}{(7.059)/_2} = 8.5"$

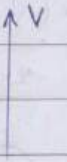


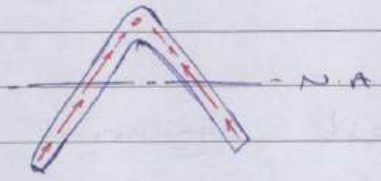












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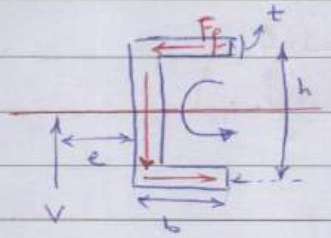
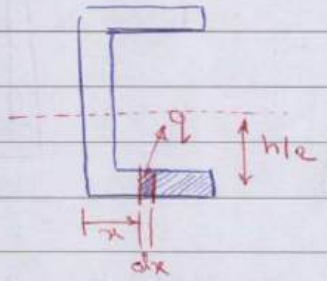


$Ve = F_f h$

$$F_f = \int_0^b q\,dx$$

$$q = \frac{Vc}{It}$$



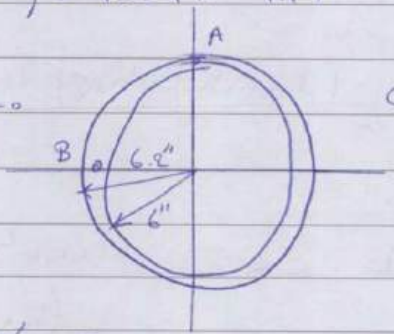
$$I = \frac{1}{12}bh^3 + 2(bt)(h/2)^2 = \frac{th^2}{2}\left(\frac{h}{6}+b\right)$$

$$q = \frac{Vc}{I} = \frac{V\frac{h}{2}(b-x)t}{\frac{th^2}{2}\left(\frac{h}{6}+b\right)} = \frac{V(b-x)}{h\left(\frac{h}{6}+b\right)}$$

$$F_q = \int_0^b q\,dx = \frac{V}{h\left(\frac{h}{6}+b\right)}\int_0^b (b-x)\,dx \Longrightarrow F_q = \frac{Vb^2}{2h\left(\frac{h}{6}+b\right)}$$

$$Ve = F_f h \longrightarrow e = \frac{b^2}{h/3+2b}$$

$$I = \pi/4\left(6.2^4 - 6^4\right) = 45.4\pi \text{ in}^4$$

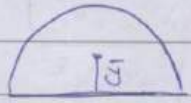


$A_{(\text{semi})}$: $q_A = 0$, $I_A = 0$

$Q_A = 0$

$$Q_B = \sum \bar{y}'A' = \frac{4(6.2)}{3\pi} \times \frac{\pi(6.2)^2}{2} - \frac{4(6)}{3\pi}\left(\frac{\pi 6^2}{2}\right)$$

$$= 14.88 \text{ in}^4$$



$$\bar{y} = \frac{4R}{3\pi}$$

$$q_B = \frac{1}{2}\frac{VQ_B}{I} = 417 \text{ lb/in}$$

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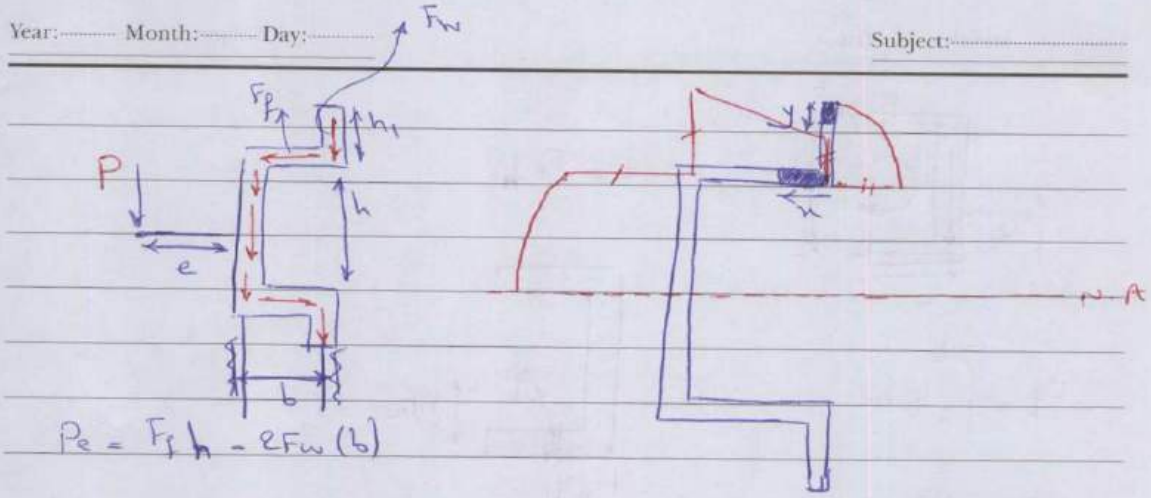

$$P_e = F_f h - e F_w (b)$$

$$I = \frac{1}{12} t(h+2h_1)^3 + 2(bt)(h/2)^2 = \frac{t}{12}\left((h+2h_1)^3 + 6bh^2\right)$$

$$Q_1 = \left(h/2 + h_1 - y/2\right) yt = t\left(h/2\, y + h_1 y - y^2/2\right)$$

$$Q_2 = \left(h/2 + h_1/2\right)(h_1 t) + h/2\, xt = t/2\left(h_1(h+h_1) + hx\right)$$

$$q_1 = \frac{VQ_1}{I} = Pt\left(h/2\, y + h_1 y - y^2/2\right)/I$$

$$q_2 = \frac{VQ_2}{I} = Pt\left(h_1(h+h_1) + hx\right)/2I$$

$$F_w = \int_0^{h_1} q_1\, dy = \frac{Pt}{I}\int_0^{h_1}(\quad)\, dy = \frac{Pt}{12I}\left(3hh_1^2 + 4h_1^3\right)$$

$$F_f = \int_0^{b} q_2\, dx = \frac{Pt}{2I}\int_0^{b}\left(h_1(h+h_1) + hx\right) dx = \frac{Pt}{2I}\left(2bh_1(h+h_1) - b^2 h\right)$$

$$P_e = F_f h - e F_w b \longrightarrow e = \frac{b(6h_1 h^2 + 9bh^2 - 8h_1^3)}{(h+2h_1)^3 + 6bh^2}$$

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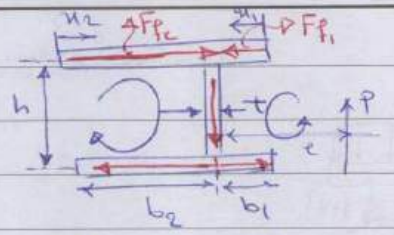
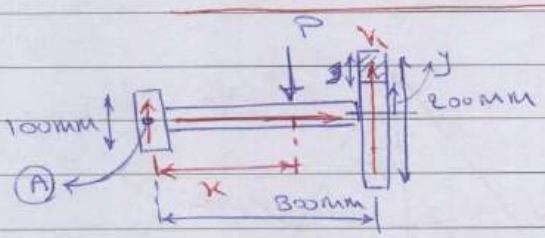
Shear:

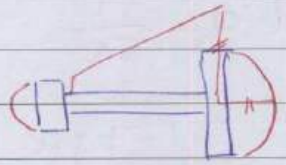


$$P_e = (F_{f_2})h - (F_{f_1})h$$

$$I = \frac{th^2}{12}\left(h + 6(b_1 + b_2)\right)$$

$$Q_1 = \frac{bI}{2} x_1 \qquad Q_2 = \frac{bI}{2} x_2 \qquad e = \frac{3(b_2^2 - b_1^2)}{h + 6(h_1 + h_2)}$$





$$\sum M_A \circlearrowleft + \cdots \Rightarrow P_e = 300 V_1$$

$$I = \tfrac{1}{12} t(0.1)^3 + \tfrac{1}{12} t(0.2)^3 = 0.75(10^{-3})\, t \ m^4$$

$$y' = y + \frac{0.1 - y}{2} = \tfrac{1}{2}(y + 0.1)$$

$$Q = y'A' = \tfrac{1}{2}(y + 0.1)(0.1 - y)\, t = t/2\,(0.01 - y^2)$$

$$q = \frac{VQ}{I} = 666.67\, P(0.01 - y^2)$$

$$V_1 = \int_{-0.1}^{0.1} q\, dy = 666.67\, P(0.01 - y^2)\, dy = 0.8889\, P$$

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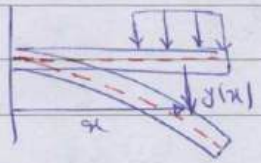
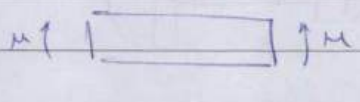
Diflection of beams / shafts خمیدگی تیرها / شفت ها

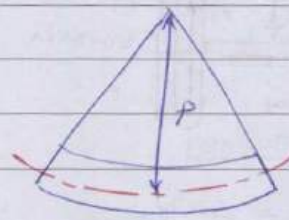
$$\begin{cases} \sigma = F/A \\ \delta\ell = \frac{FL}{AE} \end{cases}$$

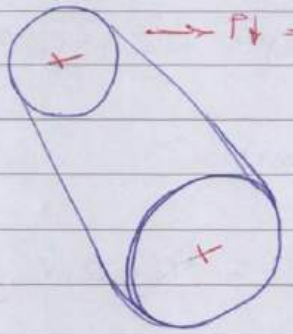


$$\begin{cases} \tau = \frac{TP}{J} \\ \phi = \frac{TL}{JG} \end{cases} \qquad \sigma = \frac{My}{I}$$

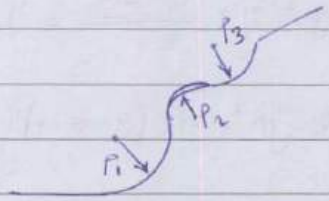
$$1/\rho = M/EI$$







$\rightarrow P\downarrow \rightarrow M\uparrow$ معمولا شفت ها باید دارای شوندگی به نوده دارای هم شود



M داریم $\rightarrow$ $1/\rho = \frac{M}{EI}$ $\rightarrow$ ρ داریم

به طور $\rightarrow$ σ داریم $\rightarrow$ $\sigma = \frac{My}{I}$

پلاستیک می رسیدند

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مثال: تسمه‌ای: 0.0625" × 0.75"

حول یک pulley دایره‌ای 12"

$1/\rho = \frac{M}{EI}$

$\Rightarrow M = \frac{EI}{\rho}$

$\sigma = \frac{Mc}{I} = \frac{EIc}{I\rho} = \frac{cE}{\rho} = 29\times10^3 \times \frac{0.0625}{12} = 20.5\ ksi$

در طول تیر ← ثابت = EI

$EI\frac{d^4y}{dx^4} = -w(x)$

انتگرال بگیر ← $EI\frac{d^3y}{dx^3} = V(x)$ ← انتگرال بگیر ← $EI\frac{d^2y}{dx^2} = M(x)$ دوستون

معادلات انتگرال خواهد (تیرهم دوستون دارد)

شرایط مرزی: $y(x=0) = 0$, $\frac{d^2y}{dx^2} = M(x=0) = 0$

شرایط مرزی: $y(x=0) = 0$, $\frac{dy}{dx}(x=0) = 0$

$M(x=0)$

$y''(x=0) = 0$

$y'''(x=0) = 0$

$V(x=0)$

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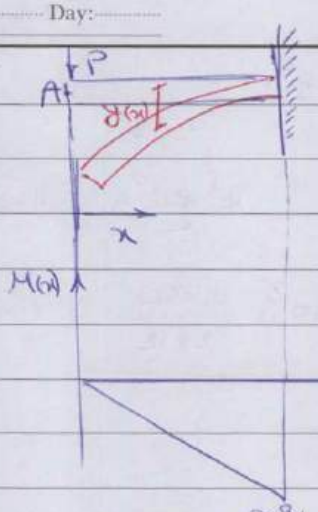


مثال

$M(x) = -Px$

$EIy'' = M(x) = -Px$

$EIy' = -Px^2/2 + c_1$

$EIy = -Px^3/6 + c_1x + c_2$

$y(x=l) = 0 \Rightarrow c_2 = -Pl^3/3$

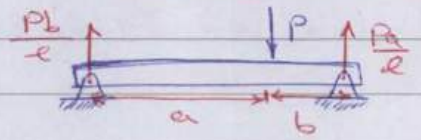
$y'(x=l) = 0 \Rightarrow c_1 = Pl^2/2$

$\Rightarrow y = \frac{P}{6EI}\left(-x^3 + 3l^2x - 2l^3\right)$

$y_{max} = y(x=0) = \frac{-Pl^3}{3EI}$ ← سوال ← most imp

$y'(x=0) = \frac{Pl^2}{2EI}$ $\quad \theta_A = \tan^{-1}\left(y'(x=0)\right)$

مثال



$EIy'' = M(x)$

$0 < x_1 < a \rightarrow EIy_1''(x_1) = M_1(x_1)$ $\quad M_1(x_1) = \frac{Pb}{l}x_1$

$EIy_1'' = \frac{Pb}{l}x_1 \Rightarrow EIy_1 = \frac{Pb}{6l}x_1^3 + c_1x_1 + c_2$

$y_1(x_1 = 0) = 0$

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$a \le x_2 \le l$ $\quad EIy_2'' = M_2(x_2)$ $\quad M_2 = \frac{Pb}{l}x_2 - P(x_2 - a)$

$\xrightarrow{b=(l-a)}$ $M_2 = Pa(1 - x_2/l)$

$EIy_2'' = Pa(1 - x_2/l) \Rightarrow EIy_2' = Pa\left(x_2 - \frac{x_2^2}{2l}\right) + c_3$

$\Rightarrow EIy_2 = Pa\left(\frac{x_2^2}{2} - \frac{x_2^3}{6l}\right) + c_3x_2 + c_4$

$y_2(x_2 = l) = 0$

$y_1(x_1 = a) = y_2(x_2 = a)$ $\quad c_1 = \frac{-Pb}{6l}(l^2 - b^2)$ $\quad c_2 = 0$

$y_1'(x_1 = a) = y_2'(x_2 = a)$

$c_3 = \frac{-Pa}{6l}(2l^2 + a^2)$ $\quad c_4 = Pa^3/6$

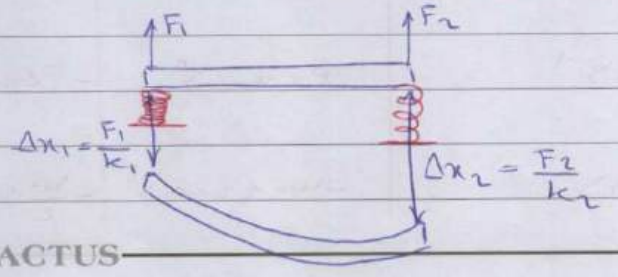
$\Rightarrow y_1 = \frac{Pb}{6EIl}\left(x_1^3 - (l^2 - b^2)x_1\right)$

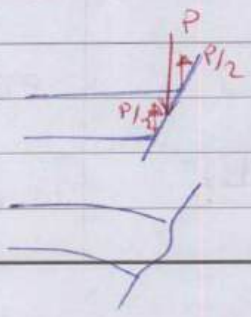
$y_2 = \frac{Pa}{6EIl}\left(3x_2^2 l - x_2^3 - (2l^2 + a^2) + a^2 l\right)$

برای یافتن MAX جزء طالب معادله ← $y_1' = 0$ ، $y_1'' = 0$ را حل کنید

معادله طالب ← معادله طالب ← $0 < x_1 < a$

$a < x_2 < l$





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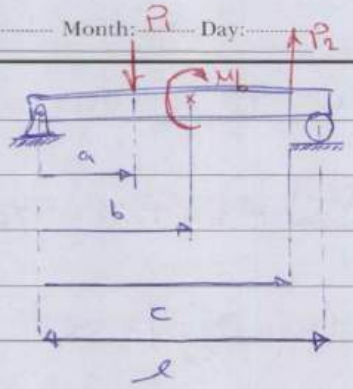


* برای سکشن $0 \le x \le a$ داریم: $M = R_1 x$

$a \le x \le b$: $M = R_1 x - P_1(x-a)$

$b \le x \le c$: $M = R_1 x - P_1(x-a) + M_b$

$c \le x \le l$: $M = R_1 x - P_1(x-a) + M_b + P_2(x-c)$

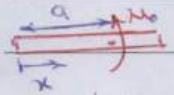
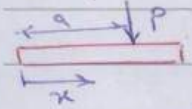
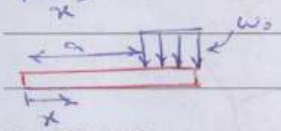
$$\langle x-a \rangle^n = \begin{cases} 0 & 0 < x < a \\ (x-a)^n & a < x < \infty \end{cases} \qquad n \ge 0$$

$$\langle x-a \rangle^n = \begin{cases} 0 & x < a \\ (x-a)^n & x \ge a \end{cases} \qquad n \ge 0$$

$$\int \langle x-a \rangle^n \, dx = \frac{\langle x-a \rangle^{n+1}}{n+1} + C \qquad n \ge 0$$

$$\int \langle x-a \rangle^n \, dx = \langle x-a \rangle^{n+1} \qquad n = -1, -2$$

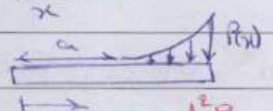
$\langle x-a \rangle^{-1}$ و $\langle x-a \rangle^{-2}$ در $x = a$ بی‌نهایت می‌شوند.

Loading	Loading function $w(x)$	Shear $V = -\int w(x)dx$	$M = \int V dx$
	$M_0 \langle x-a \rangle^{-2}$	$-M_0 \langle x-a \rangle^{-1}$	$-M_0 \langle x-a \rangle^{0}$
	$P \langle x-a \rangle^{-1}$	$-P \langle x-a \rangle^{0}$	$-P \langle x-a \rangle^{1}$
	$w_0 \langle x-a \rangle^{0}$	$-w_0 \langle x-a \rangle^{1}$	$-\frac{w_0}{2} \langle x-a \rangle^{2}$

Year: Month: Day: $w(x)$ $V(x)$ Subject: $M(x)$

$m = \frac{dP}{dx}$ شیب

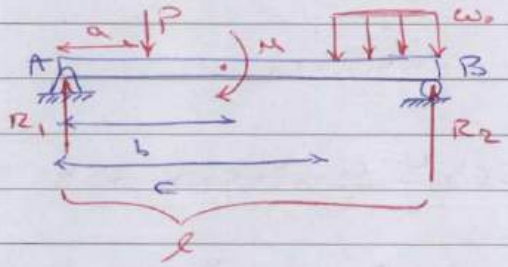
$m\langle x-a\rangle^1 \quad -m/2\langle x-a\rangle^2 \quad -m/6\langle x-a\rangle^3$



$c/2\langle x-a\rangle^2 \quad -\frac{c}{6}\langle x-a\rangle^3 \quad -\frac{c}{24}\langle x-a\rangle^4$

$c = \frac{d^2P}{dx^2}$

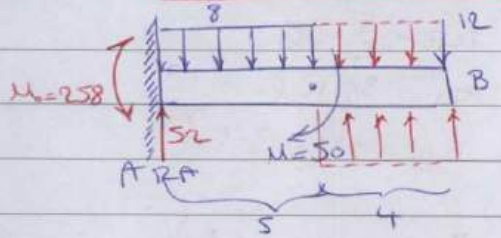
مثال:



$$w(x) = -R_1\langle x-0\rangle^{-1} + P\langle x-a\rangle^{-1} - M_0\langle x-b\rangle^{-2} + w_0\langle x-c\rangle^0$$

اگر عکس العمل R_1 را در ابتدا نمیدانیم (یا نشناسا) در ثابت اول انتگرال ضرایب R_1 ثابت اول ضرایب R_1 ، ثابت دوم ضرایب R_2 و ... ثابت های دوم و ... ثابت های

مثال:



$$w(x) = -52\langle x-0\rangle^{-1} + 258\langle x-0\rangle^{-2} + 8\langle x-0\rangle^0 - 8\langle x-5\rangle^0 - 50\langle x-5\rangle^{-2} = EI\,y^{(4)}_{(x)}$$

* ثابت R_A و M_0 را در $M(x)$ قرار میدهیم پس از آن انتگرال گرفته و ثابت را محاسبه می‌کنیم

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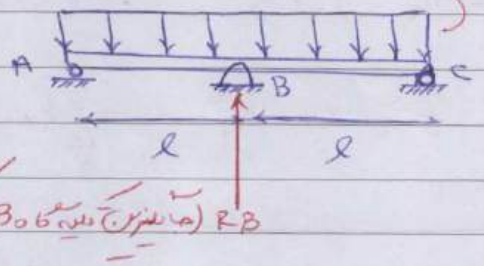


$EI y^{(4)} = w(x) = -P_0 + R_B \langle x-\ell \rangle^{-1}$

$EI y^{(3)} = -P_0 x^0 + R_B \langle x-\ell \rangle^{0} + c_1$

$EI y'' = \frac{-P_0}{2} x^2 + R_B \langle x-\ell \rangle + c_1 x + c_2$

$EI y' = \frac{-P_0}{6} x^3 + \frac{R_B}{2} \langle x-\ell \rangle^2 + \frac{c_1}{2} x^2 + c_2 x + c_3$

$EI y = \frac{-P_0}{24} x^4 + \frac{R_B}{6} \langle x-\ell \rangle^3 + \frac{c_1}{6} x^3 + \frac{c_2}{2} x^2 + c_3 x + c_4$

c_1, c_2, c_3, c_4, R_B شرط ضریب‌ها را تعیین کنید:

$y(0) = y(\ell) = y(2\ell) = 0$

$EI y''(0) = EI y''(2\ell) = 0$

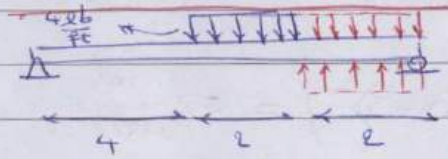
$c_4 = c_2 = 0$

$R_B = \frac{5}{4} P_0 \ell$

$c_1 = \frac{3}{8} P_0 \ell$

$c_3 = -\frac{1}{48} P_0 \ell^3$

$\Rightarrow y(x) = \frac{-P_0}{48 EI} \left(2x^4 - 8\ell x^3 - 10\ell \langle x-\ell \rangle^3 + \ell^3 x \right)$



$EI y^{(4)} = -4 \langle x-4 \rangle^0 + 4 \langle x-6 \rangle^0$

$EI y^{(3)} = -4 \langle x-4 \rangle^1 + 4 \langle x-6 \rangle^1 + c_1$

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$$EIy'' = -2\langle x-4\rangle^2 + 2\langle x-6\rangle^2 + C_1x + C_2$$

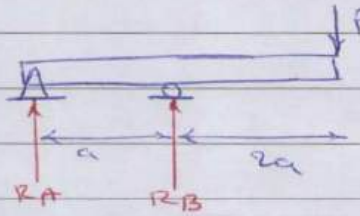
$y''(0) = 0$, $y''(8) = 0$ $\Rightarrow$ $C_1 = 8$, $C_2 = 0$

$$\rightarrow EIy'' = -2\langle x-4\rangle^2 + 2\langle x-6\rangle^2 + 8x$$

$$\rightarrow EIy' = -2/3\langle x-4\rangle^3 + 2/3\langle x-6\rangle^3 + 8/2\,x^2 + C_3$$

$$\rightarrow EIy = -1/6\langle x-4\rangle^4 + 1/6\langle x-6\rangle^4 + \frac{1}{2}x^3 + C_3x + C_4$$

$y(0) = y(12) = 0 \Rightarrow C_4 = 0$, $C_3 = -27$



$RA = -2P$

$RB = 3P$

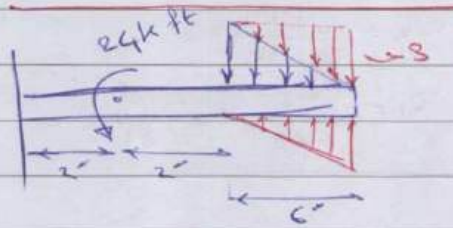
$$EIy'' = -2Px + 3P\langle x-a\rangle^1$$

$$EIy' = -Px^2 + 3P/2\langle x-a\rangle^2 + C_3$$

$$EIy = \frac{-Px^3}{3} + \frac{P}{2}\langle x-a\rangle^3 + C_3x + C_4$$

$y(0) = y(a) = 0 \rightarrow C_4 = 0$

$C_3 = Pa^2/8$



dP/dx

$$EIy''' = -24\langle x-2\rangle^{-2} - 3\langle x-4\rangle^0 + 3/6\langle x-4\rangle^1$$

$$EIy'' = -24\langle x-2\rangle^{-1} - 3\langle x-4\rangle^1 + \frac{3}{12}\langle x-4\rangle^2 + C_1$$

$EIy''(10) = 0 \Rightarrow C_1 = 9^k$!!! ...

$$EIy'' = -24\langle x-2\rangle^0 - 3/2\langle x-4\rangle^2 + 1/12\langle x-4\rangle^3 + 9x + C_2$$

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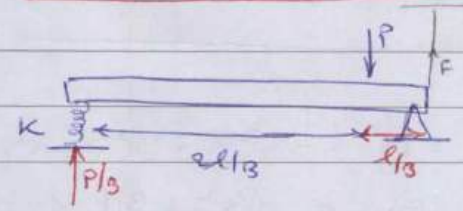
$EI\,y''(0) = 0 \Rightarrow c_2 = -30k$

$EI\theta(x) = -24\langle x-2\rangle^1 - 1/2\langle x-4\rangle^3 + \frac{1}{48}\langle x-4\rangle^4 + 9/2\,x^2 + 30x + c_3$

$\theta(6) = 0 \Rightarrow c_3 = 0$

$EIy = -12\langle x-2\rangle^2 - \frac{1}{8}\langle x-4\rangle^4 + \frac{1}{240}\langle x-4\rangle^5 + 3/2\,x^3 - 15x^2 + c_4$

$y(0) = 0 \Rightarrow c_4 = 0$



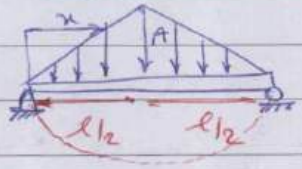
$\delta = \frac{Fl}{AE} = y(l)$

$EIy'' = Px/3 - P\langle x - 2l/3\rangle^1$

$EIy' = Px^2/6 - P/2\langle x - 2l/3\rangle^2 + c_1$

$EIy = \frac{Px^3}{18} - P/6\langle x - 2l/3\rangle^3 + c_1x + c_2$

$\begin{cases} y(6) = -\frac{P}{3k} \\ y(l) = 0 \end{cases} \longrightarrow \begin{cases} c_2 = -\frac{PEI}{3k} \\ c_1 = \frac{PEI}{3kl} - \frac{4Pl^2}{81} \end{cases}$



$Al/4$

$0 < x < l/2 \qquad EIy'' = Rx - \frac{Ax^3}{3l}$

$EIy = \frac{A}{12EI}\left(l/2\,x^3 - \frac{x^5}{5l} + \frac{5l^3}{16}x\right)$

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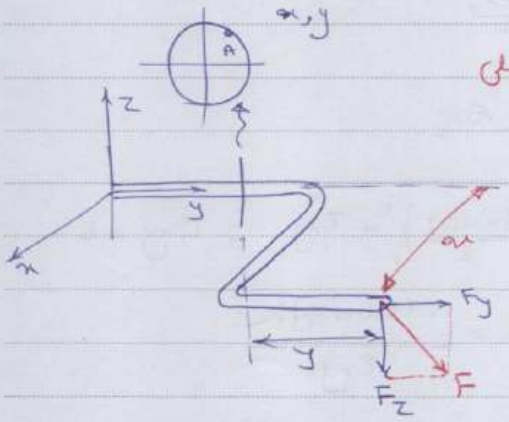
Stress Transformation. (تبدیل تنش)

$$\sigma_A = \begin{pmatrix} \sigma_x & \tau_{xy} \\ \tau_{yx} & \sigma_y \end{pmatrix} = \begin{pmatrix} \sigma_1^{(1)} & 0 \\ 0 & \sigma_2^{(2)} \end{pmatrix} = \begin{pmatrix} \frac{\sigma_x+\sigma_y}{2} & \tau_{max} \\ \tau_{max} & \frac{\sigma_x+\sigma_y}{2} \end{pmatrix}$$

σ_x, σ_y σ_x, σ_y اصلی $\bar{\sigma}_x, \bar{\sigma}_y$

* تنش های اصلی

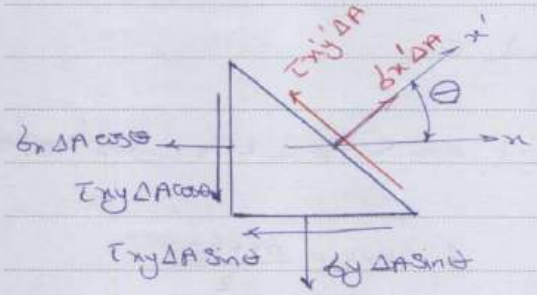


$F_z(x)$ → نیروی محوری در صفحه

$F_z(y)$ → برشی در صفحه

F_y → محوری در صفحه

F_z → برشی خارج صفحه



$$\sum F_{x'} = 0 \Rightarrow \sigma_{x'}\Delta A - \tau_{xy}\Delta A \sin\theta\cos\theta + \cdots$$

$$\cdots - \sigma_y \Delta A \sin^2\theta - \tau_{xy}\Delta A \cos\theta \sin\theta = 0$$

$$\sigma_{x'} = \sigma_x \cos^2\theta + \sigma_y \sin^2\theta + \tau_{xy}(2\sin\theta\cos\theta)$$

$$\sigma_{x'} = \frac{\sigma_x+\sigma_y}{2} + \frac{\sigma_x-\sigma_y}{2}\cos 2\theta + \tau_{xy}\sin 2\theta$$

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$$\sum F_{y'} = 0 \Rightarrow \tau_{x'y'} = -\left(\frac{\sigma_x - \sigma_y}{2}\right)\sin 2\theta - \tau_{xy}\cos 2\theta$$

$$\theta \rightarrow \theta + \pi/2 \qquad \sigma_{y'} = \frac{\sigma_x + \sigma_y}{2} - \frac{\sigma_x - \sigma_y}{2}\cos 2\theta - \tau_{xy}\sin 2\theta$$

$$\frac{d\gamma_{ax'}}{d\theta} = 0 \Rightarrow \tan 2\theta_p = \frac{\tau_{xy}}{\frac{\sigma_x - \sigma_y}{2}}$$

$2\theta_{P_1}$ "(180)" $2\theta_{P_2} \Rightarrow \theta_{P_1} \perp \theta_{P_2}$

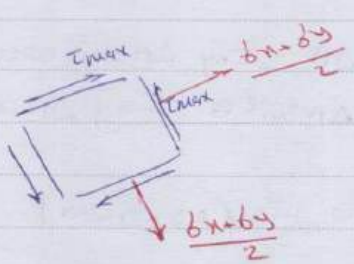
$$\sigma^{(1),(2)} = \frac{\sigma_x + \sigma_y}{2} \pm \sqrt{\left(\frac{\sigma_x - \sigma_y}{2}\right)^2 + \tau_{xy}^2} \quad ; \tau_{xy} = 0$$

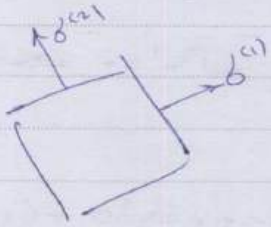
$$\frac{d\tau_{x'y'}}{d\theta} = 0 \Rightarrow \tan 2\theta_s = \frac{-(\sigma_x - \sigma_y)/2}{\tau_{xy}}$$

$2\theta_{S_1}$ "(180)" $2\theta_{S_2} \Rightarrow \theta_{S_1} \perp \theta_{S_2}$

$(\tan 2\theta_p) * (\tan 2\theta_s) = -1 \Rightarrow 2\theta_p \perp 2\theta_s \Rightarrow \theta_p$ "(45)" θ_s

$$\tau_{max} = \sqrt{\left(\frac{\sigma_x - \sigma_y}{2}\right)^2 + \tau_{xy}^2} \quad ; \sigma_{ave} = \frac{\sigma_x + \sigma_y}{2}$$

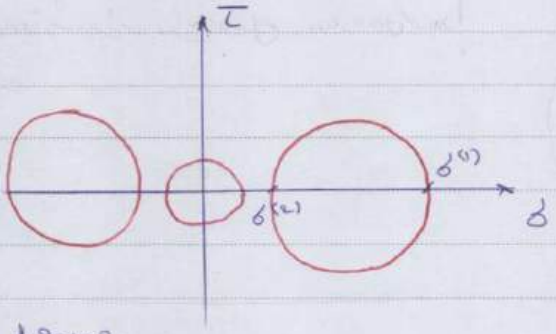




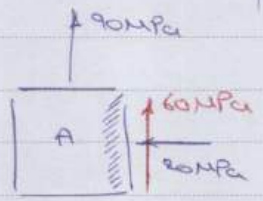
Subject:

Year. Month. Date. ( )

* واحده آنای صفحه *



مثال:



$\sigma_x = -20$

$\sigma_y = 90$

$\tau = 60$

$$\tan 2\theta_p = \frac{\tau_{xy}}{\frac{\sigma_x - \sigma_y}{2}} \longrightarrow 2\theta_{p_2} = -47.49$$

$$\Rightarrow \theta_{p_2} = -23.7$$

$$\Rightarrow \theta_{p_1} = 66.3$$

$$\sigma^{(1),(2)} = \frac{\sigma_x + \sigma_y}{2} \pm \sqrt{\left(\frac{\sigma_x - \sigma_y}{2}\right)^2 + \tau_{xy}^2}$$

$$\Rightarrow \begin{cases} \sigma^{(1)} = 116\,MPa \\ \sigma^{(2)} = -46.4\,MPa \end{cases}$$ متناظر با θ_{p_1} و θ_{p_2} است

$$\tan 2\theta_s = \frac{-\left(\frac{\sigma_x - \sigma_y}{2}\right)}{\tau_{xy}} \qquad 2\theta_s = 42.5$$

$$\theta_{s_1} = 21.3$$

$$\theta_{s_2} = 111.3$$

$$\tau_{max} = \sqrt{\left(\frac{\sigma_x - \sigma_y}{2}\right)^2 + \tau_{xy}^2} = 81.4$$

$$\sigma = \begin{pmatrix} -20 & 60 \\ 60 & 90 \end{pmatrix} = \begin{pmatrix} 116 & 0 \\ 0 & -46.4 \end{pmatrix} + \begin{pmatrix} 35 & 81.4 \\ 81.4 & 35 \end{pmatrix}$$

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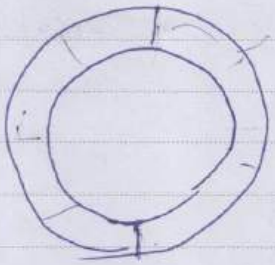
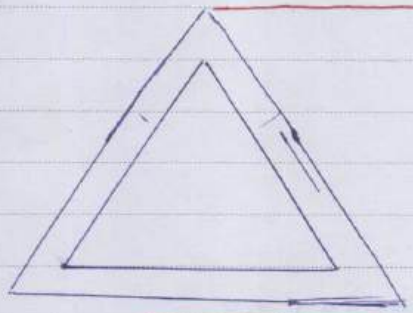
* حلقه کامل ، حلقه شیاردار ، رابط کامل ، رباط مفصل شده





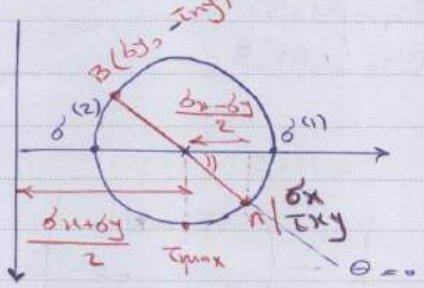
$$\left(\sigma x' - \frac{\sigma x + \sigma y}{2}\right) = \left(\frac{\sigma x - \sigma y}{2}\cos 2\theta + \tau xy \sin 2\theta\right)$$

$$(\tau x'y') = \left(-\frac{\sigma x - \sigma y}{2}\sin 2\theta + \tau xy \cos 2\theta\right)$$

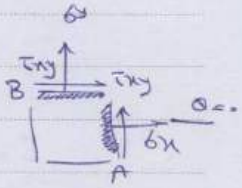
$$\left(\sigma x' - \frac{\sigma x + \sigma y}{2}\right)^2 + \tau x'y'^2 = \left(\frac{\sigma x - \sigma y}{2}\right)^2 + \tau xy^2$$

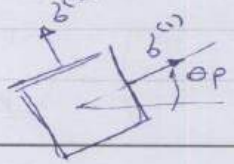
$$(x - a)^2 + (y - b)^2 = R^2$$

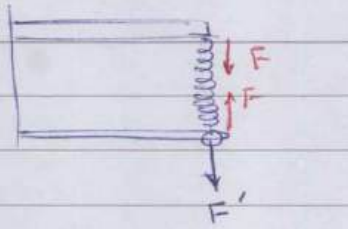
$$R = \sqrt{\left(\frac{\sigma x - \sigma y}{2}\right)^2 + \tau xy^2}$$

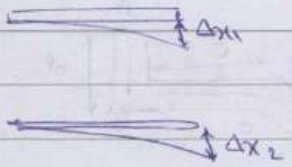


$$\tan 2\theta p = \frac{\tau xy}{\frac{\sigma x - \sigma y}{2}}$$

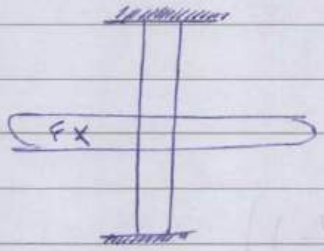


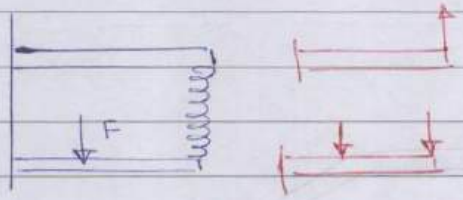


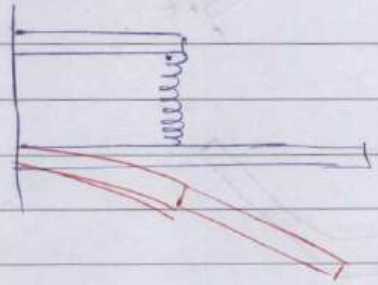


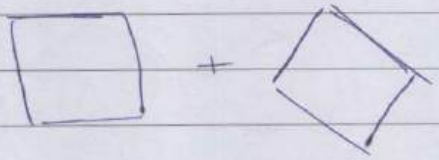


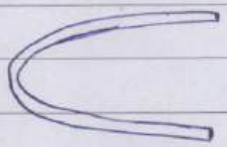
$|\Delta x_2 - \Delta x_1| = F/k$

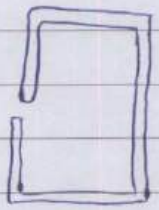












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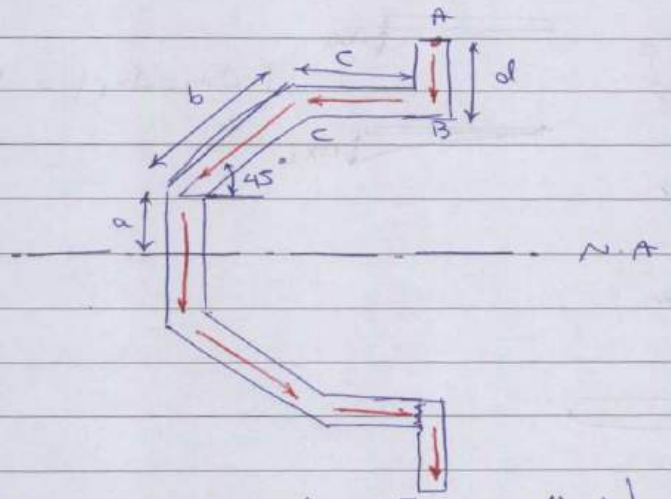
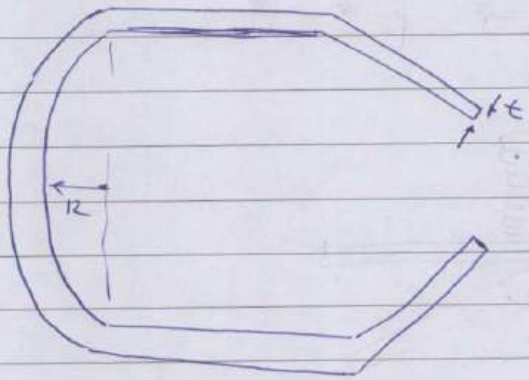


$A-B: \tau_{xy} = \frac{V}{It}\left(Ut\left(a + b\sqrt{2}/2 + d - \frac{U}{2}\right)\right)$

$B-C: \tau_{xz} =$



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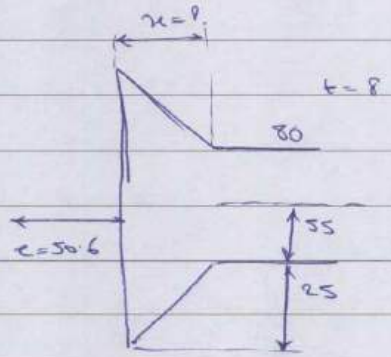
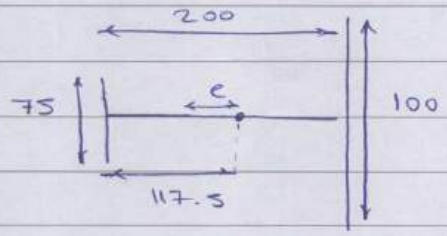


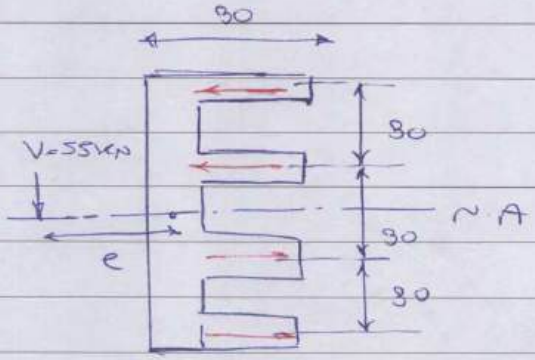
t=6



$I_y = 1.902 \times 10^6$

$I_z = 1.526 \times 10^6$

(e=81.6 maybe)



$I_z = 0.9333 \times 10^6$

$e = 9.12$